

# Fracture Response of an Adhesive-Bonded Thermoplastic Composite at Low Temperature

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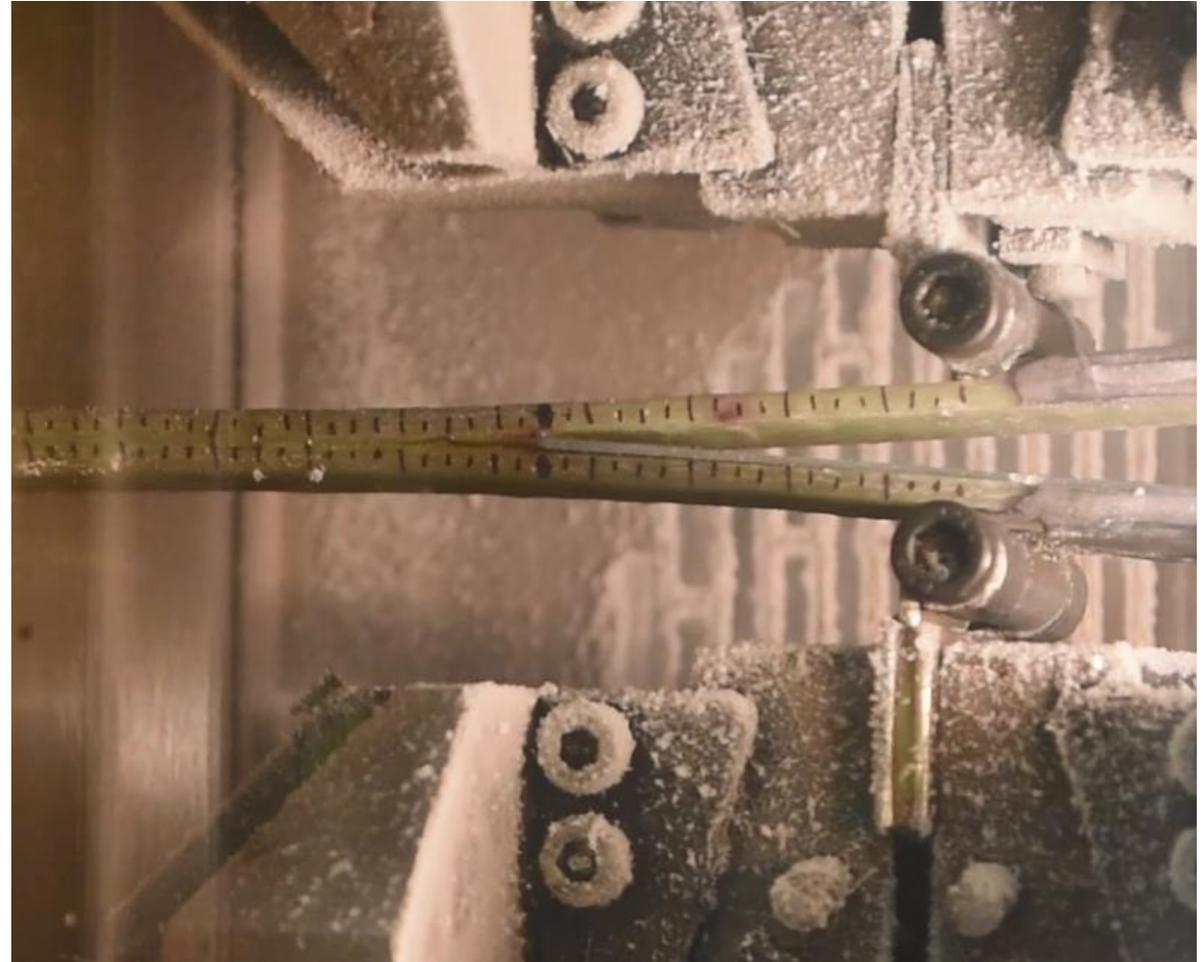
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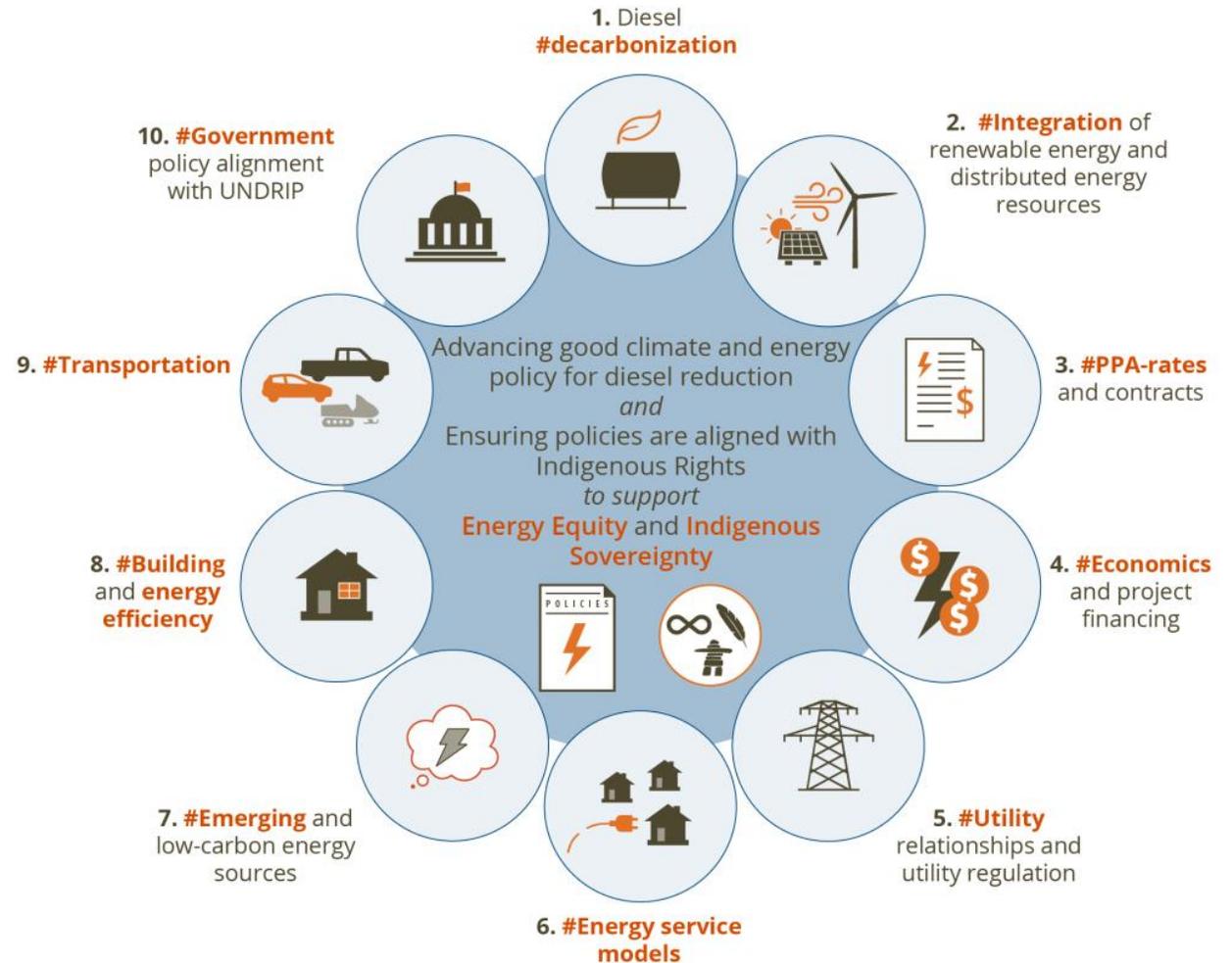
# Agenda

- Background
- Introduction
- Sample Preparation
- Experimental Method
- Experimental Results
- Conclusions



# Background – Motivation for Work

- Clean and sustainable energy sources are critical for remote communities
- Micro-generating wind turbines present a viable alternative to diesel generators
- Harsh environments impose a high demand on wind turbine blades



*Roadmap for supporting energy projects and promoting diesel reduction in indigenous communities<sup>1</sup>.*

# Background – Turbine Blade Manufacturing

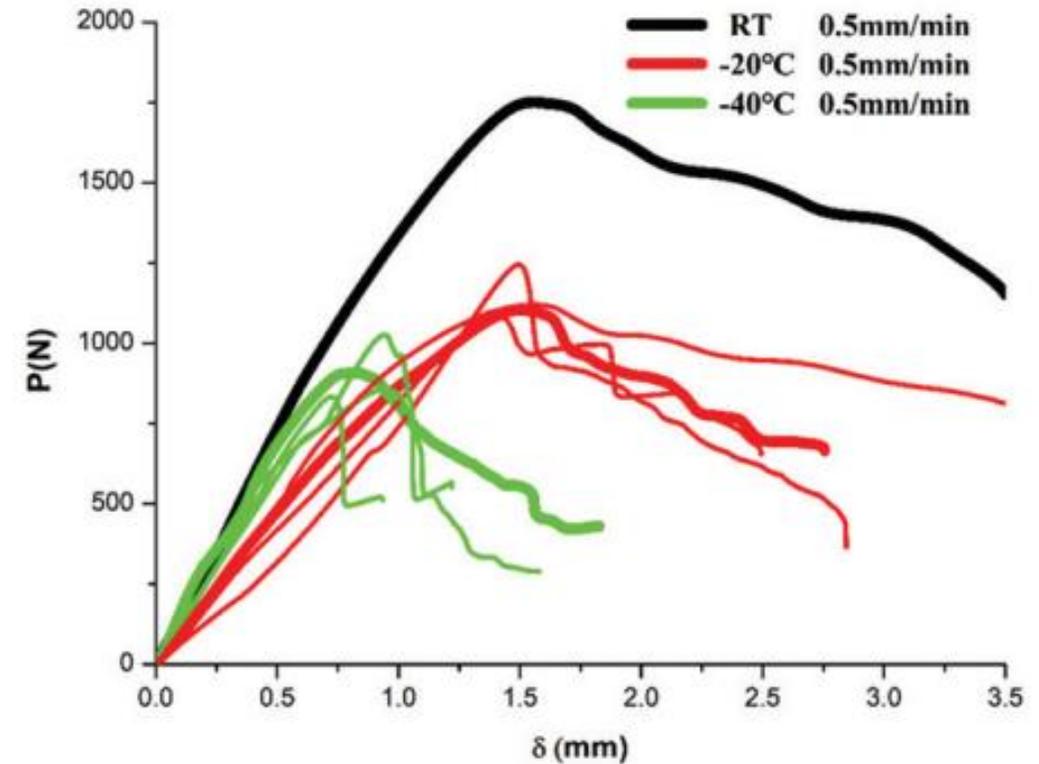
- Our group has developed a resin infusion process for blade shells
- Resin used is a reactive polymethyl methacrylate (PMMA)
- Turbine blade assembly is composed of two adhesively-bonded composite blade shells
- **Thermoplastic composites are more difficult to bond than thermoset composites**



*Adhesive-bonded micro-generating wind turbine blade.*

# Introduction – Thermoplastic Adhesives

- Murray et al. successfully bonded thermoplastic composites with methyl methacrylate (MMA) adhesives<sup>2</sup>
- Thermoplastic adhesive testing with variable temperature has been conducted by Jia et al.<sup>3</sup>
- To the author's knowledge, low temperature testing of methacrylate adhesive with thermoplastic substrate has not been reported



*Jia et al. demonstrated that temperature has a drastic effect on polyurethane (thermoplastic) adhesive performance at quasi-static loading conditions<sup>3</sup>. Adhesive used to bond steel substrate.*

**Objective: Characterizing the temperature and bond-line thickness effect on Mode-I & Mode-II fracture response of a thermoplastic adhesive**

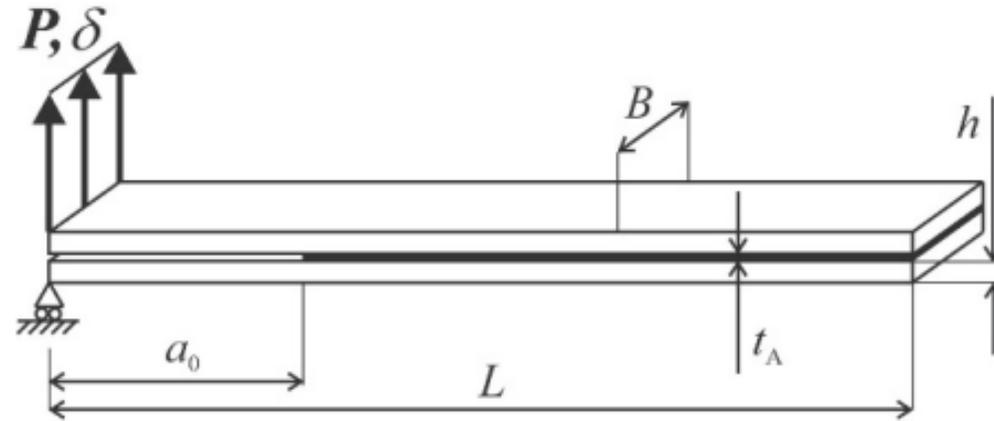
# Sample Preparation – Substrate

- **Material system:** Elium<sup>®</sup> 188 XO PMMA resin (Arkema) reinforced with glass fiber unidirectional non-crimp fabric (UD-NCF) (SAERTEX<sup>®</sup>)
- Substrates underwent abrasion and degreasing surface treatment on mold-side surface prior to bonding
- Substrates bonded with FIT30-45 MMA adhesive (Bostik) with bond-line thicknesses in the range 0.2mm to 1.0mm



*Cross-sectional image of thermoplastic fiberglass panel.  $[0]_4$  stacking sequence with  $\approx 48\%$  FVF.*

# Sample Preparation – DCB Bonding

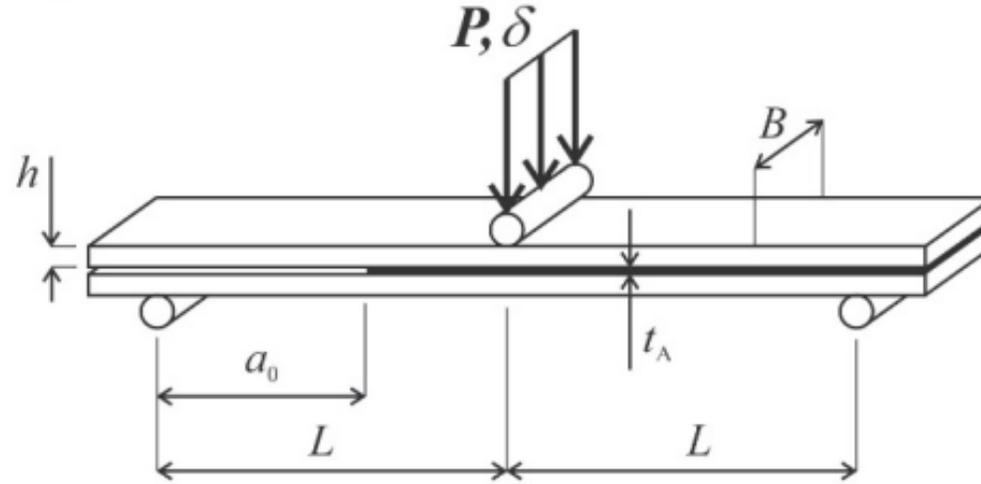


DCB specimen schematic<sup>4</sup>.

DCB specimen dimensions.

| Shim thickness (mm) | Length $L$ (mm) | Width $B$ (mm) | Bond-line Thickness $t_A$ (mm $\pm$ STV) | Pre-Crack; Shim length $a_0$ (mm $\pm$ STV) | Total Sample Thickness $2h$ (mm $\pm$ STV) | Average Surface Roughness ( $\mu\text{m}$ $\pm$ STV) |
|---------------------|-----------------|----------------|--|---|--|--|
| 0.25                | 140             | 25.4           | $0.3268 \pm 0.04905$                     | $45.52 \pm 0.4417$                          | $6.008 \pm 0.08802$                        | $0.764 \pm 0.0469$                                   |
| 0.635               |                 |                | $0.7225 \pm 0.1247$                      | $45.41 \pm 0.4704$                          | $6.348 \pm 0.06547$                        |  |
| 0.8                 |                 |                | $0.8073 \pm 0.07316$                     | $45.89 \pm 0.4164$                          | $6.507 \pm 0.06178$                        |  |

# Sample Preparation – ENF Bonding

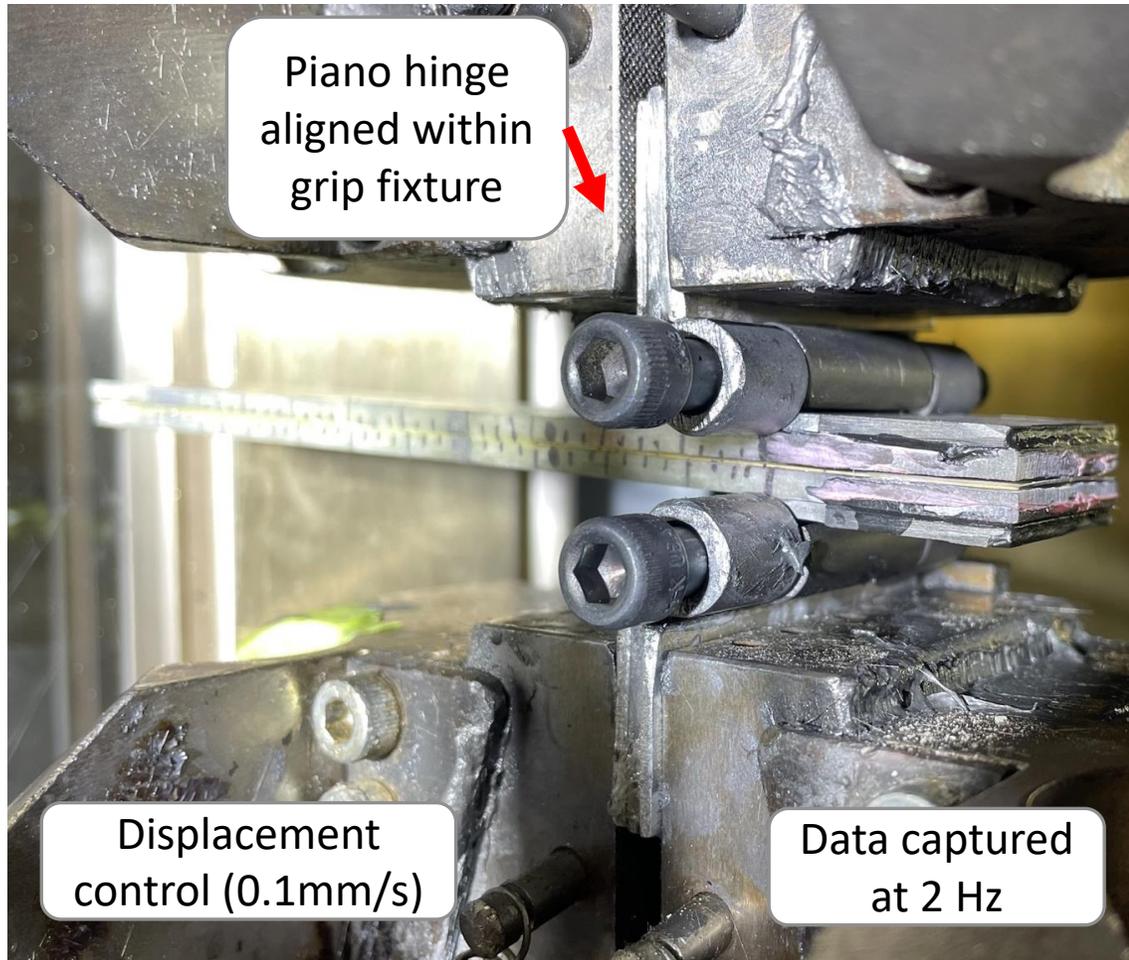


ENF specimen schematic<sup>4</sup>.

ENF specimen dimensions.

| Shim thickness (mm) | Mid-Span Length $L$ (mm) | Width $B$ (mm) | Bond-line Thickness $t_A$ (mm $\pm$ STV) | Pre-Crack; Shim length $a_0$ (mm $\pm$ STV) | Total Sample Thickness $2h$ (mm $\pm$ STV) | Average Surface Roughness ( $\mu\text{m}$ $\pm$ STV) |
|---------------------|--------------------------|----------------|--|---|--|--|
| 0.25                | 75                       | 25.4           | 0.3041 $\pm$ 0.04639                     | 51.41 $\pm$ 0.2635                          | 6.147 $\pm$ 0.1819                         | 0.764 $\pm$ 0.0469                                   |
| 0.475               |                          |                | 0.6052 $\pm$ 0.02707                     | 52.68 $\pm$ 0.5493                          | 6.429 $\pm$ 0.08791                        |  |
| 0.675               |                          |                | 0.7211 $\pm$ 0.06853                     | 52.46 $\pm$ 0.3609                          | 6.480 $\pm$ 0.06961                        |  |

# Experimental Method – Fixture Setup



*Piano hinge fixture setup for DCB testing.*



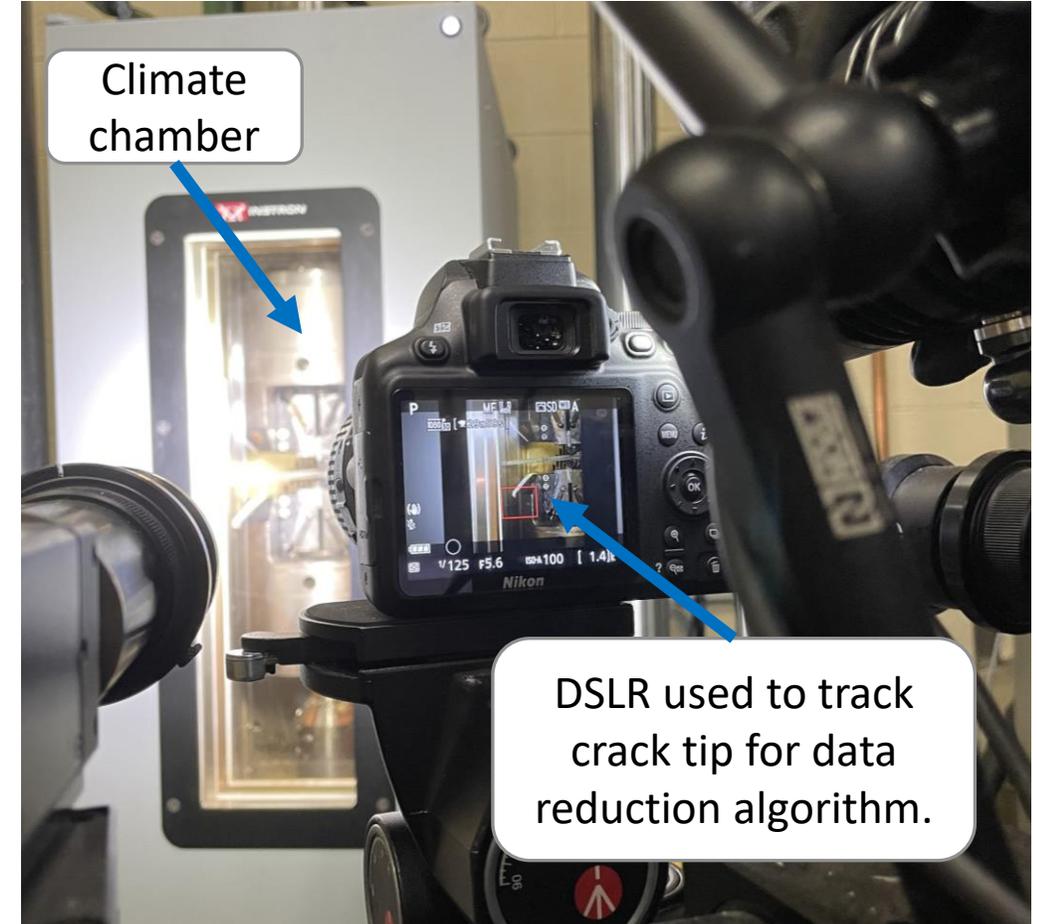
*3 Point Bend (3PB) fixture setup for ENF testing.*

# Experimental Method – LT Test Setup

- Samples tested using MTS 810 servo-hydraulic test frame
- Climate chamber with LN2 canister used for LT tests
- Cooling rate of  $-6.5^{\circ}\text{C}/\text{minute}$  followed by conditioning for 5 minutes at set temperature of  $-40^{\circ}\text{C}$



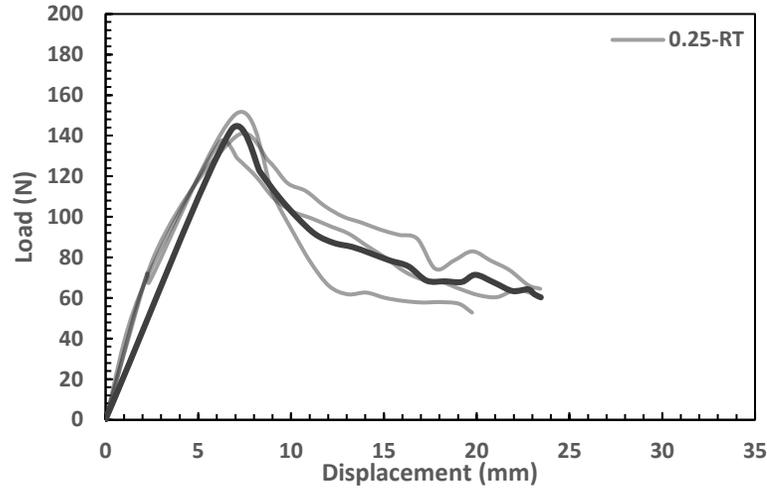
*LN2 canister used to cool climate chamber for LT tests.*



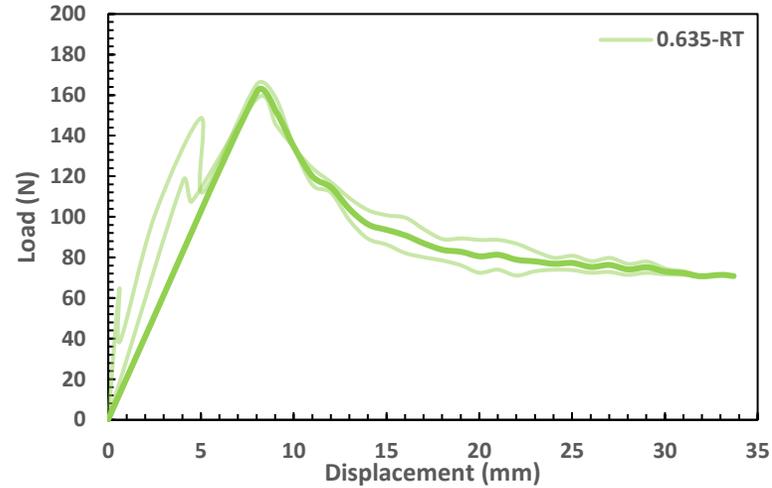
*Sample enclosed within climate chamber; crack front still visible through glass with DSLR.*

# Experimental Results – Mode-I

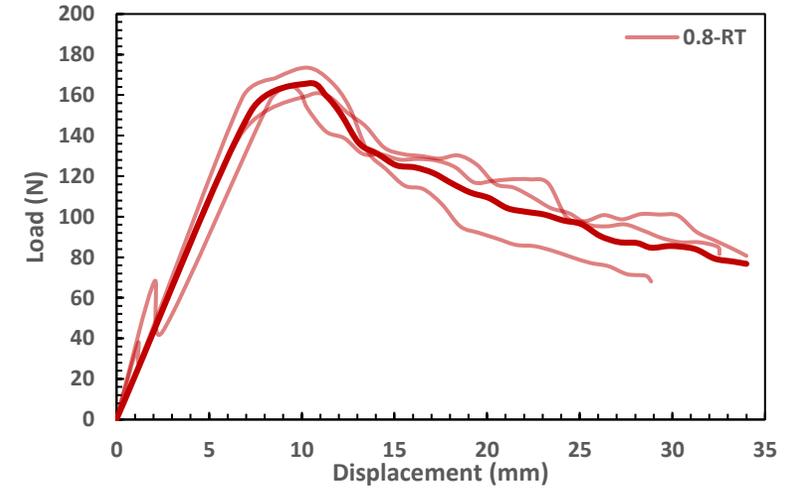
0.25mm



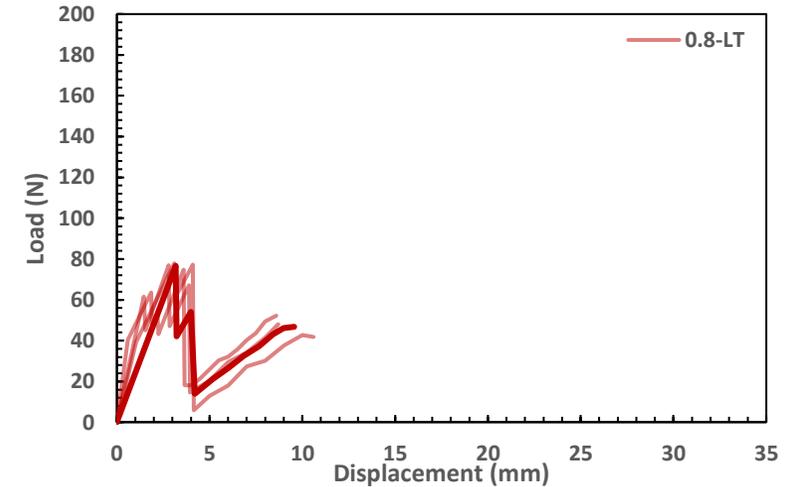
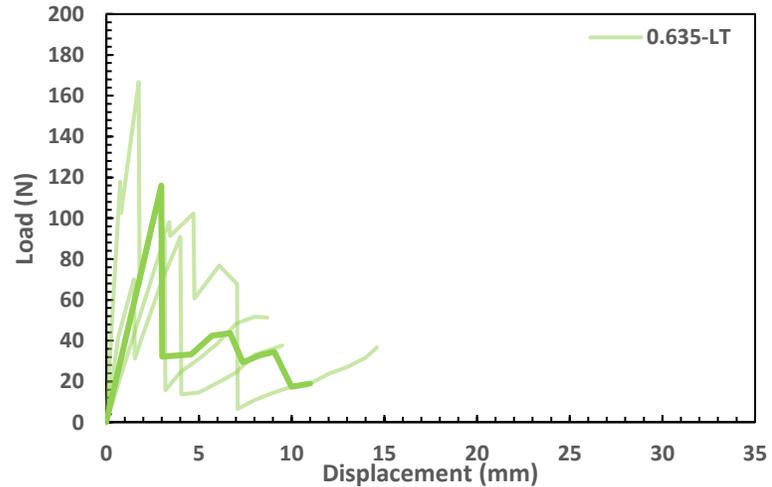
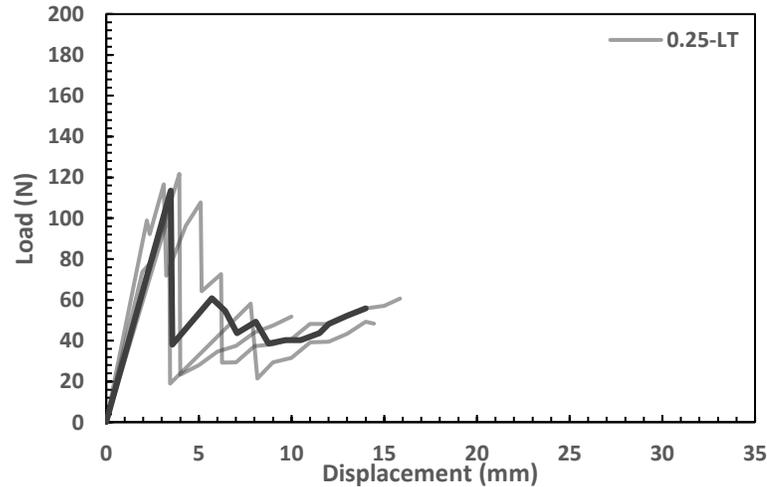
0.635mm



0.8mm



RT



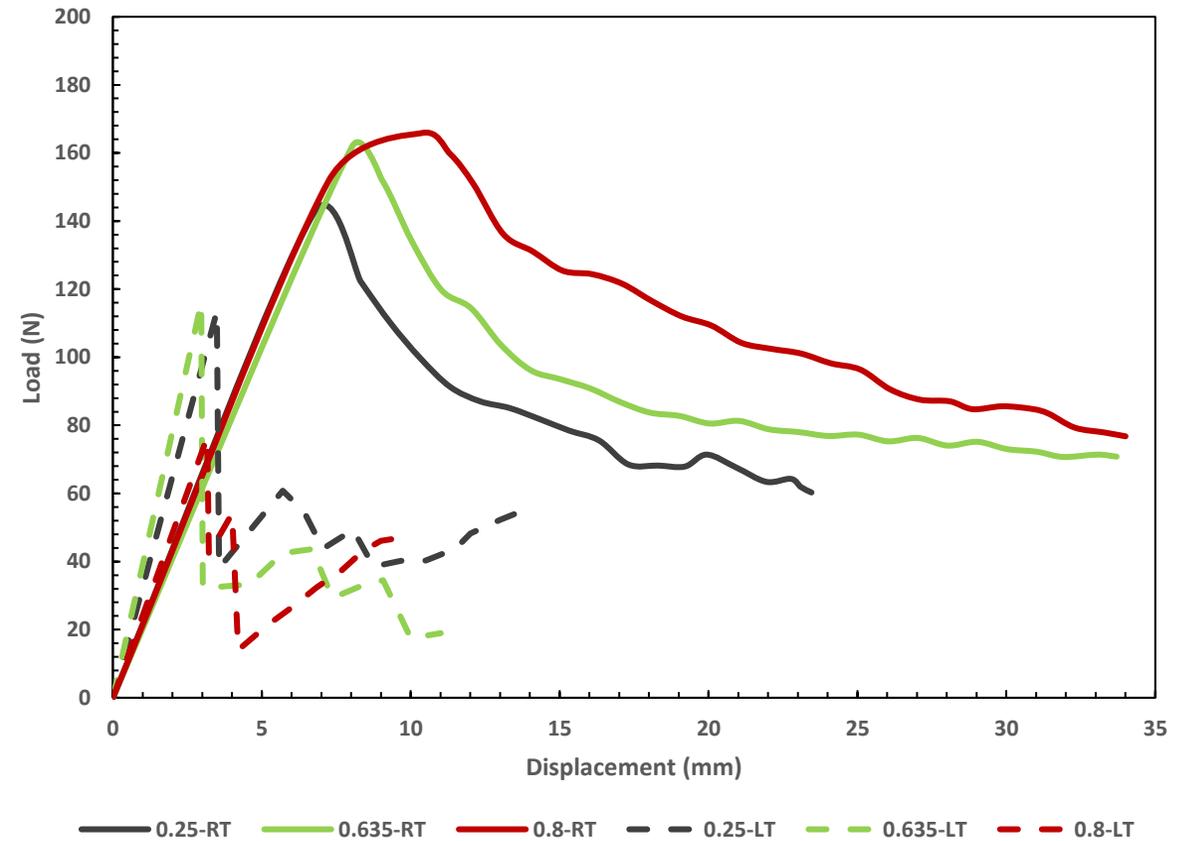
LT

# Experimental Results – Mode-I Fracture Toughness

- *Modified Beam Theory (MBT) Method (from ASTM D5528)* used to determine mode-I strain energy release rate ( $G_{IC}$ )<sup>5</sup> for RT-DCB samples
- LT-DCB  $G_{IC}$  calculations incomplete due to errors in elastic curve

Below:  $G_{IC}$  calculations using MBT method for RT-DCB specimens.

| Bond-line Thickness (mm) | $G_{IC}$ (N/mm) |
|--------------------------|-----------------|
| 0.25                     | 0.929           |
| 0.635                    | 1.231           |
| 0.8                      | 1.632           |



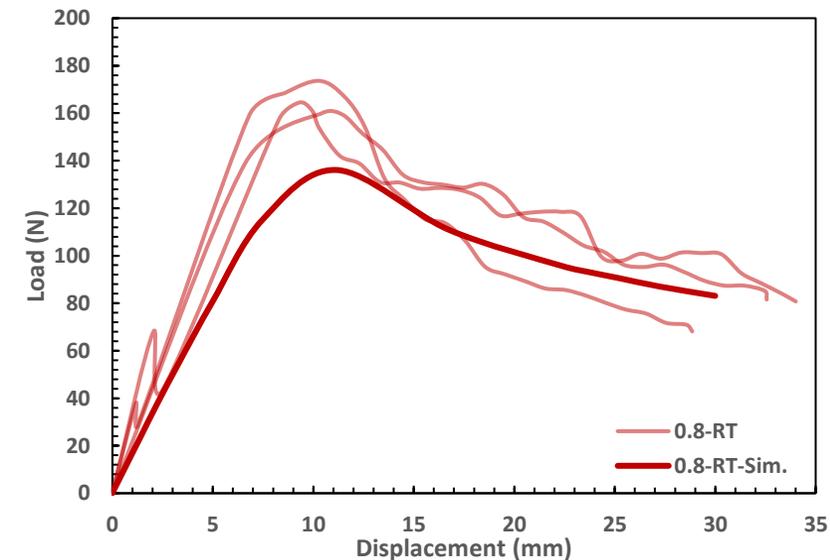
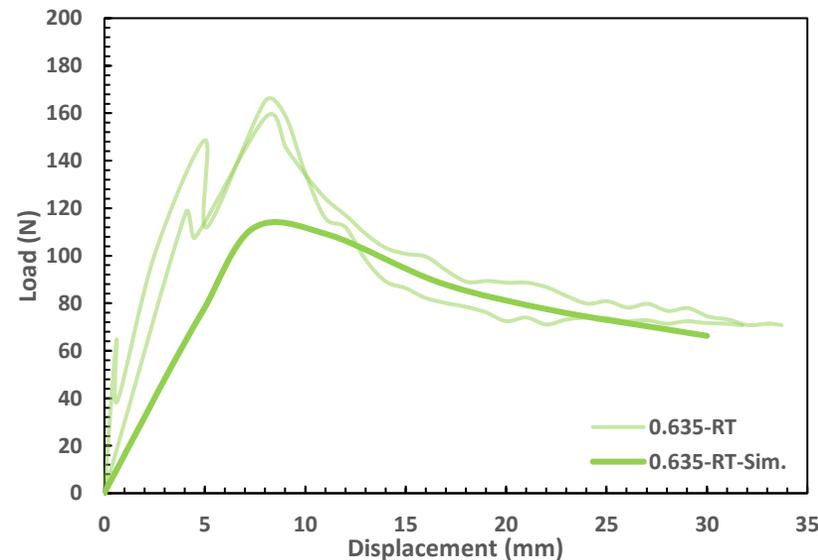
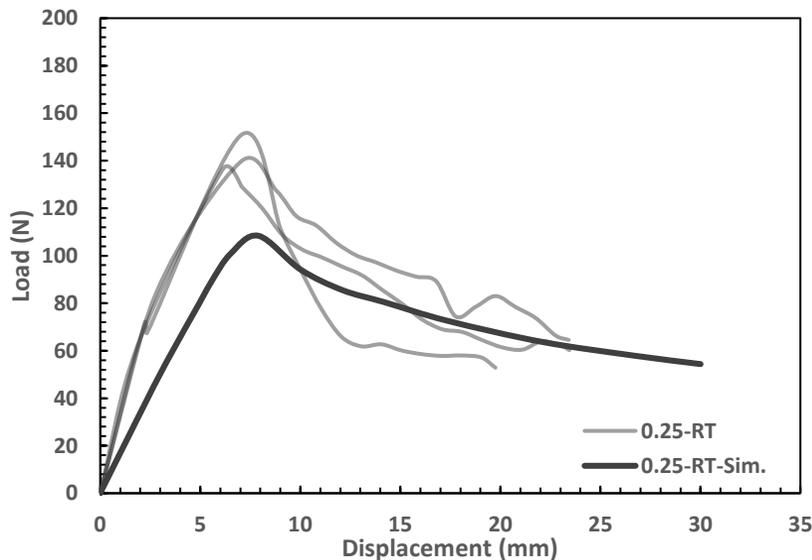
Overlay of average Load-Displacement curves for DCB tests at RT and LT.

# Numerical Model – Mode-I RT

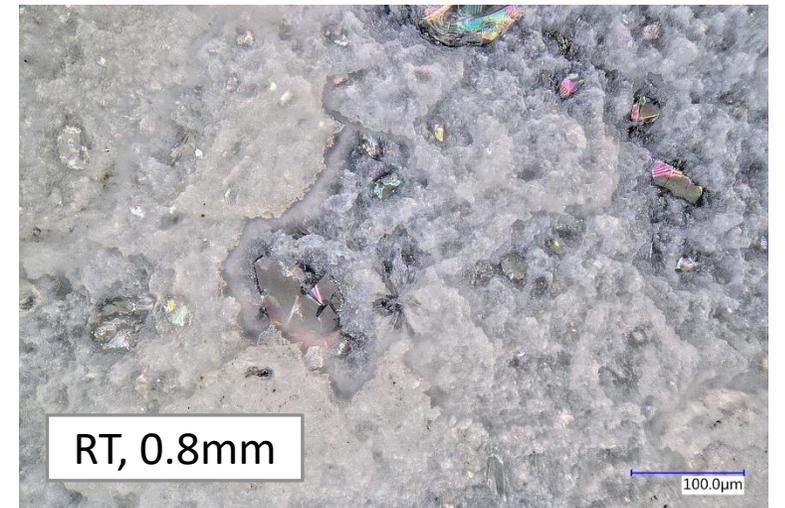
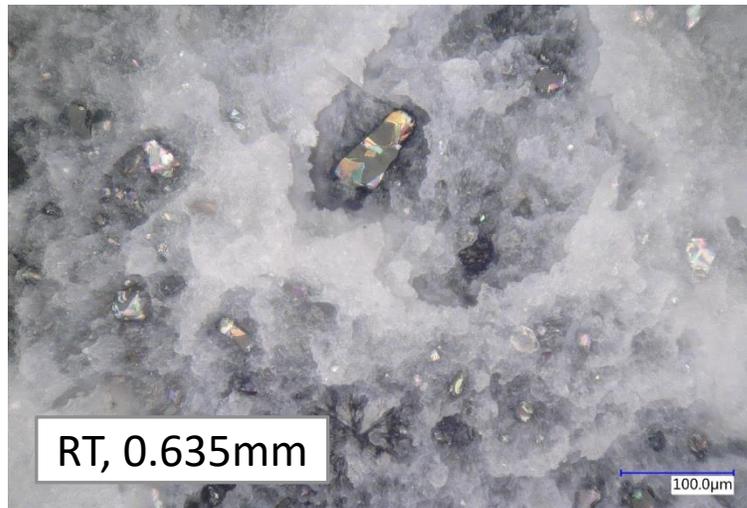
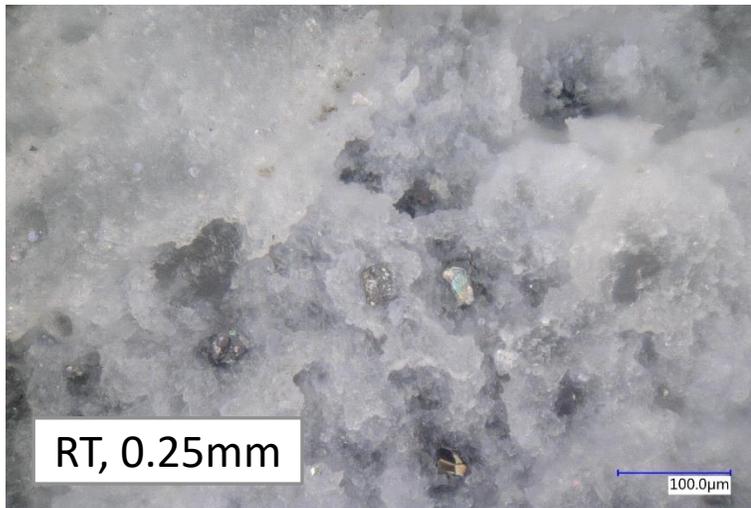
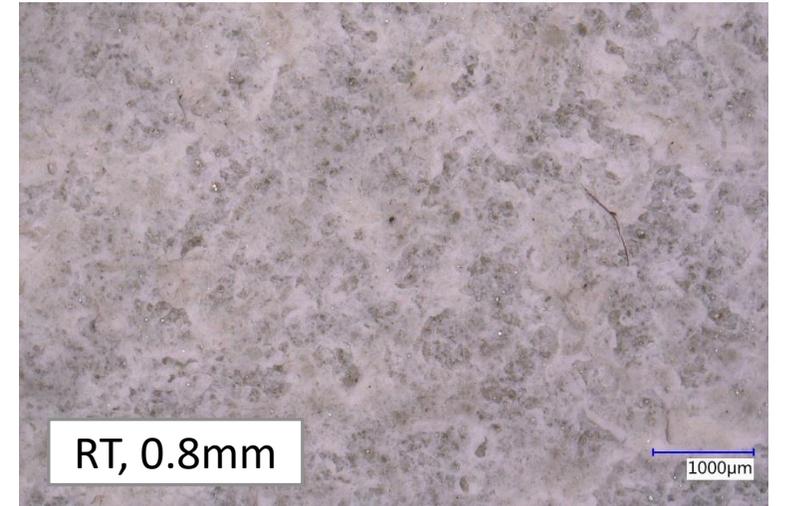
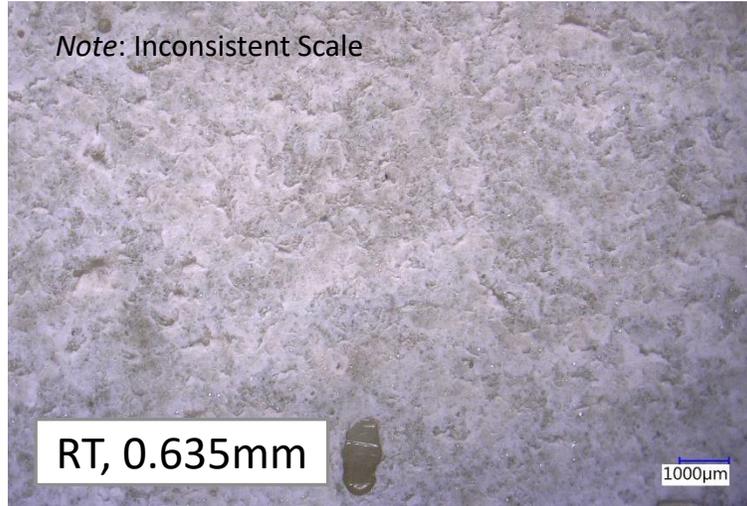
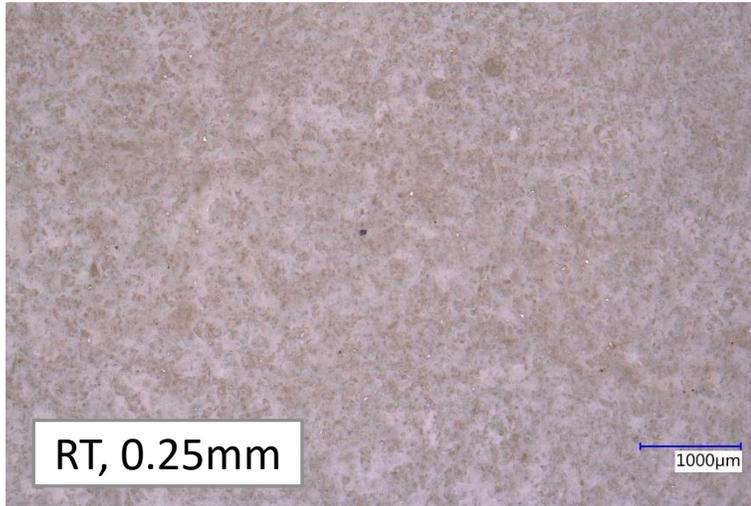
- RT-DCB experiments simulated in ABAQUS using cohesive elements to determine TSL parameters
- $G_{IC}$  from experimental results used as fracture energy input
- Discrepancies in elastic response may be due to process errors during sample bonding

*Bilinear traction separation law parameters for ABAQUS simulations.*

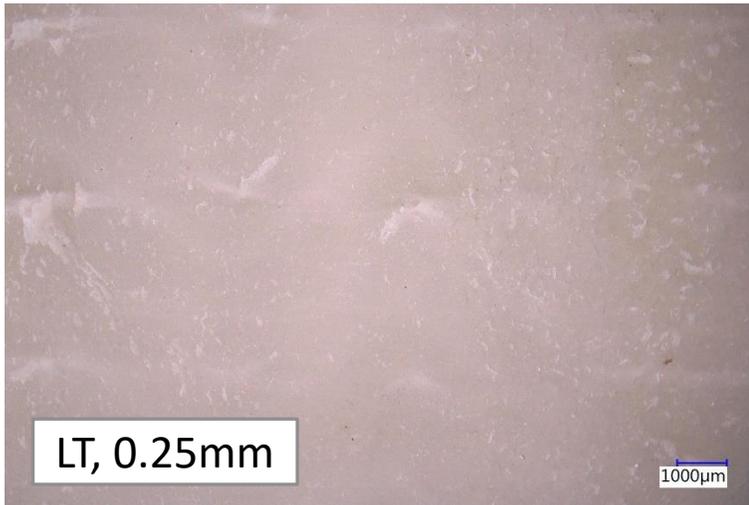
| Bond-line Thickness (mm) | Initial Stiffness (MPa) | Peak Traction (MPa) | Fracture Energy (MPa mm) |
|--------------------------|-------------------------|---------------------|--------------------------|
| 0.25                     | 1000                    | 15                  | 0.929                    |
| 0.635                    | 500                     | 15                  | 1.23                     |
| 0.8                      | 2000                    | 15                  | 1.63                     |



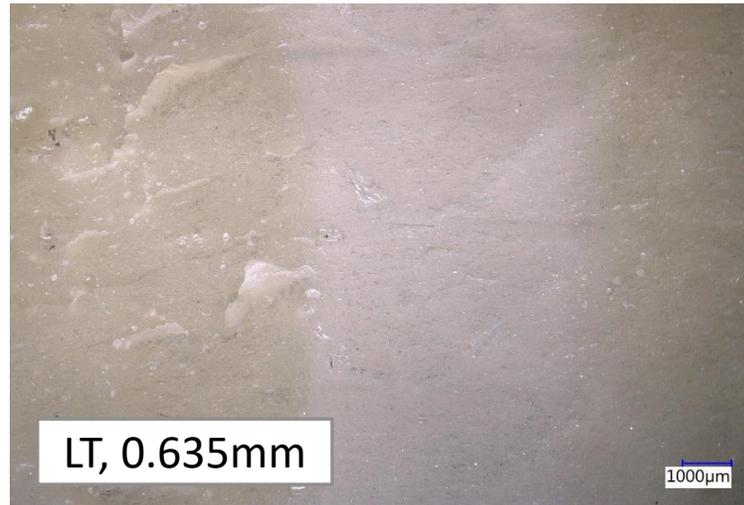
# Fracture Analysis – Mode-I RT



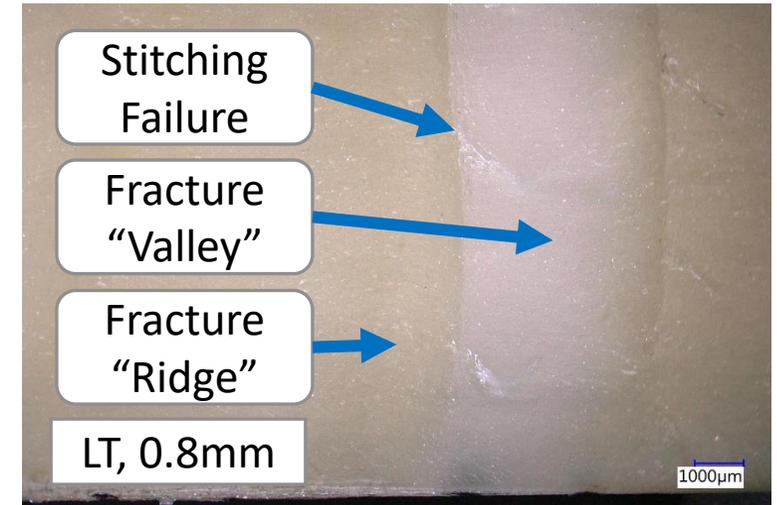
# Fracture Analysis – Mode-I LT



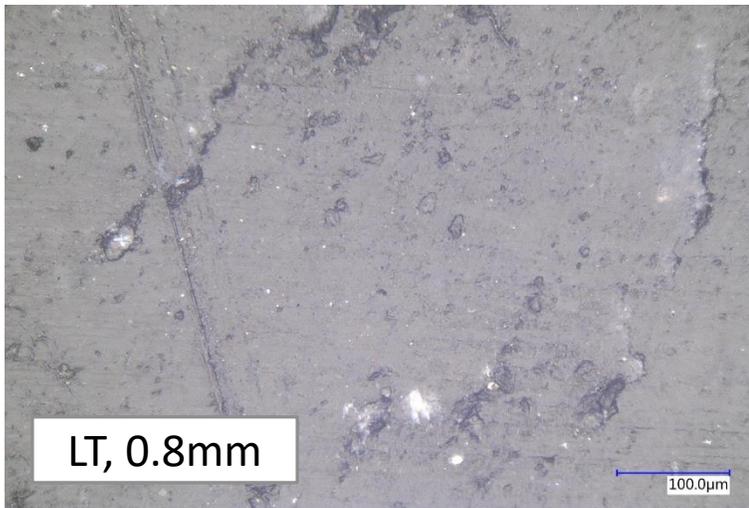
*LT fracture surface.*



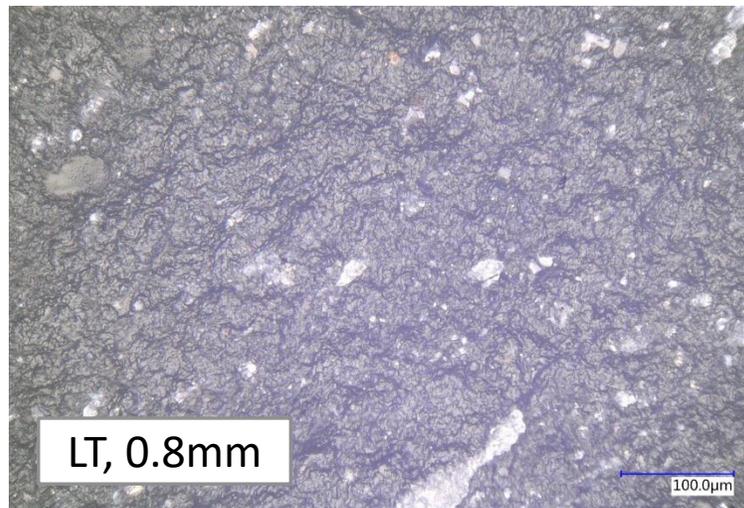
*LT fracture ridge/valley.*



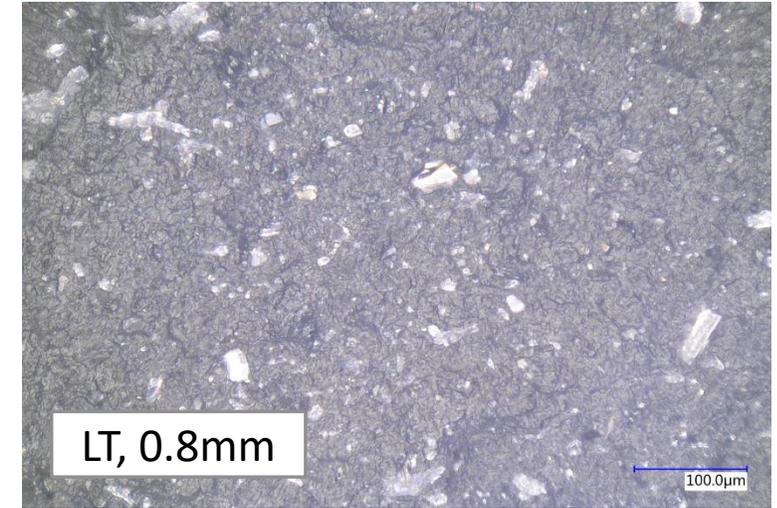
*LT fracture ridge/valley with exposed stitching.*



*Unbonded composite at 500x magnification.*



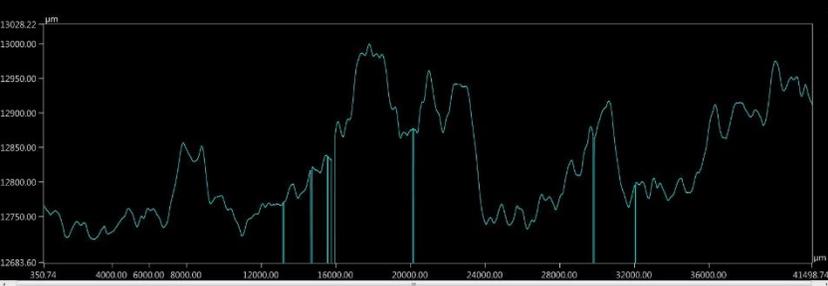
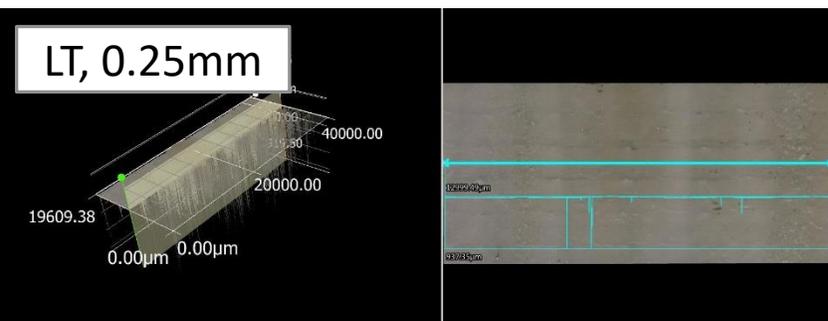
*LT fracture ridge at 500x magnification.*



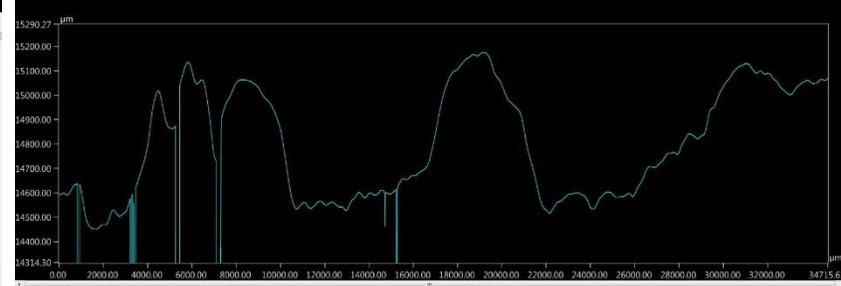
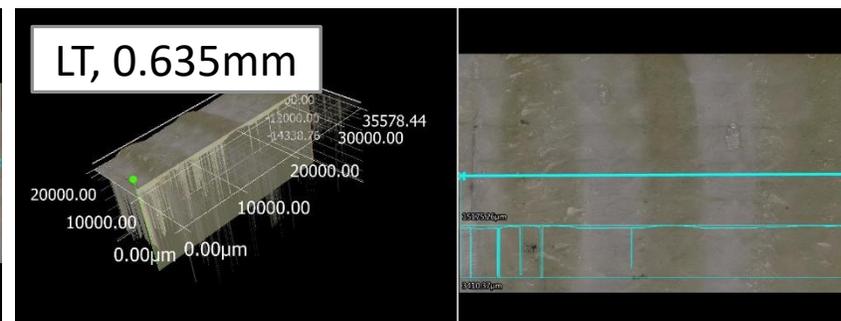
*LT fracture valley at 500x magnification.*

# Fracture Analysis – Mode-I LT

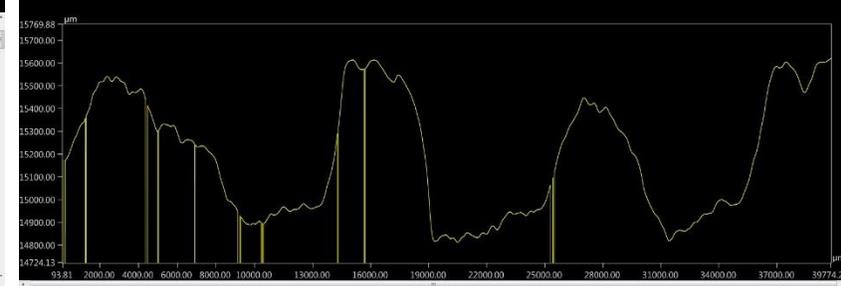
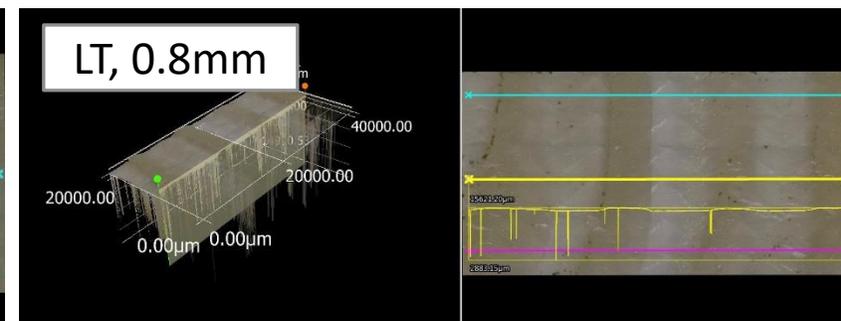
- LT-DCB samples characterized by oscillatory “groove” patterns (valleys and ridges)
- Residual stresses caused by specimen cooling may have increased crack-path instability (T-stress theory)<sup>6</sup>
- 3D depth display was used to characterize surface roughness for varying adhesive thicknesses



Rz: 201 $\mu$ m



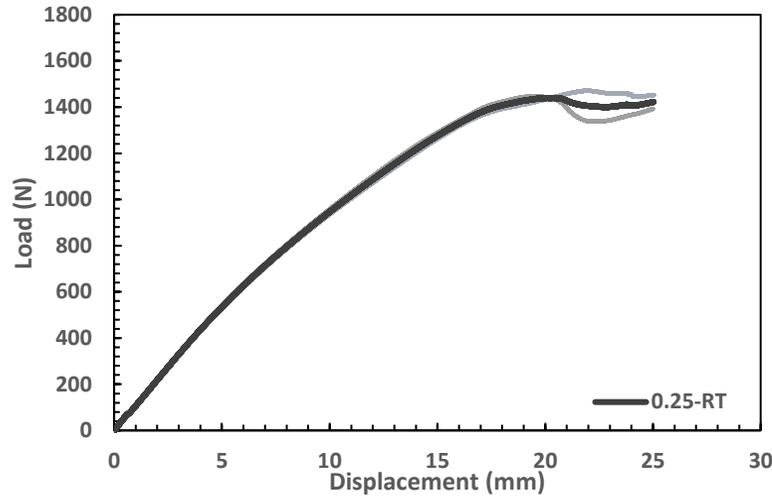
Rz: 646 $\mu$ m



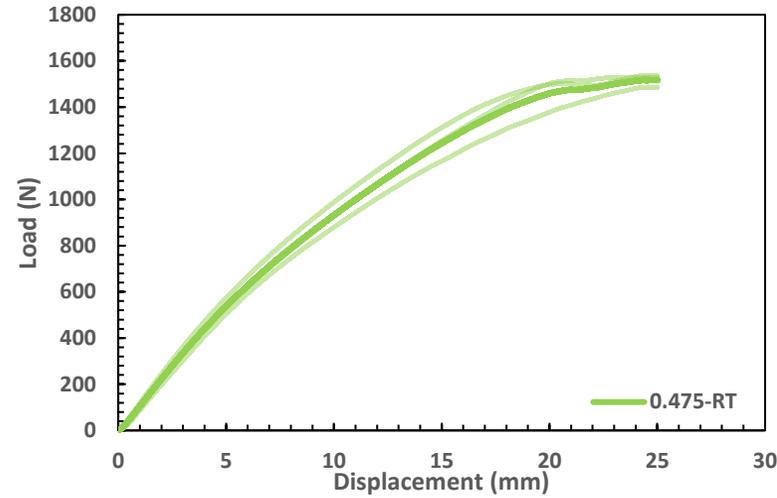
Rz: 681 $\mu$ m

# Experimental Results – Mode-II

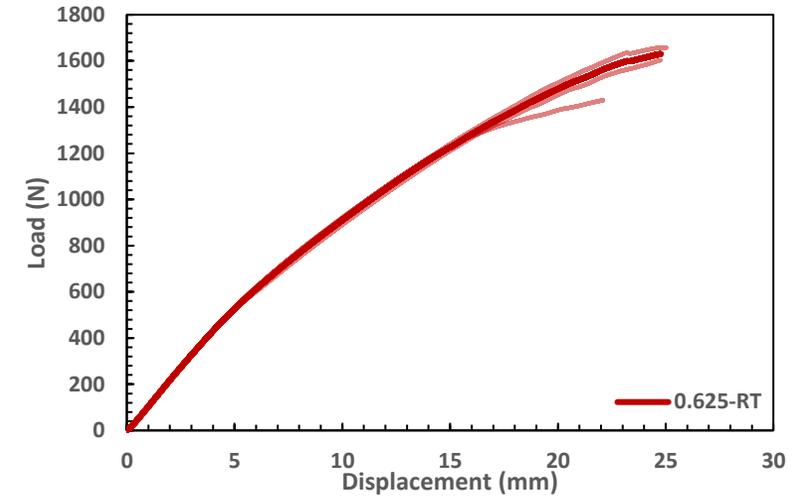
0.25mm



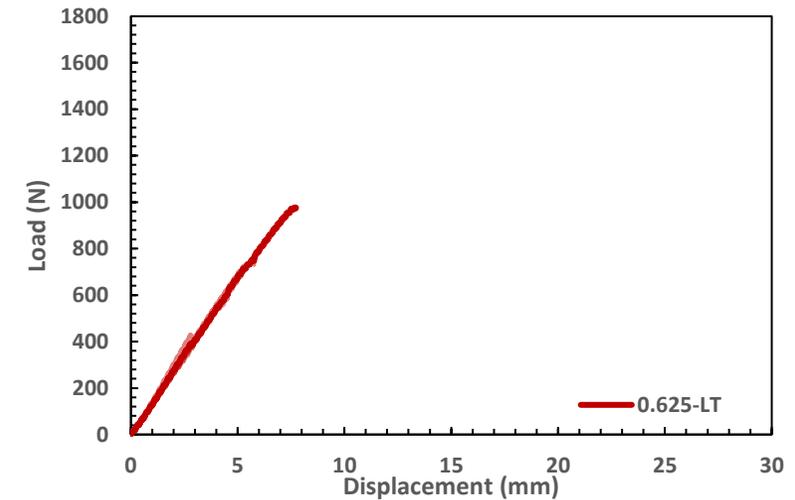
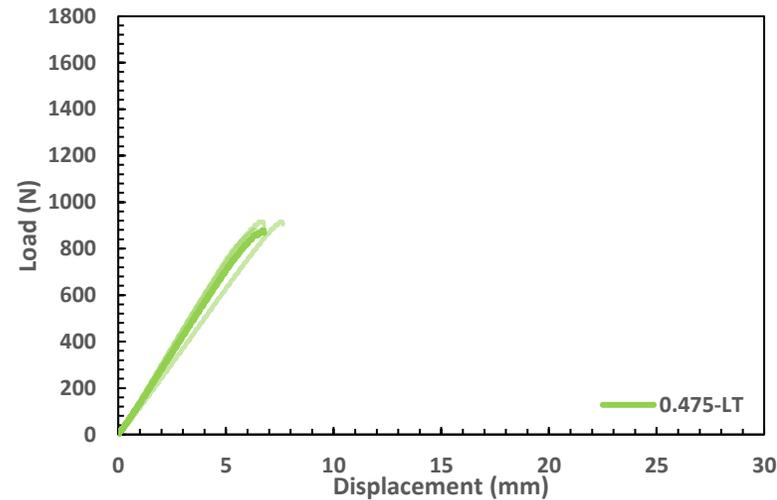
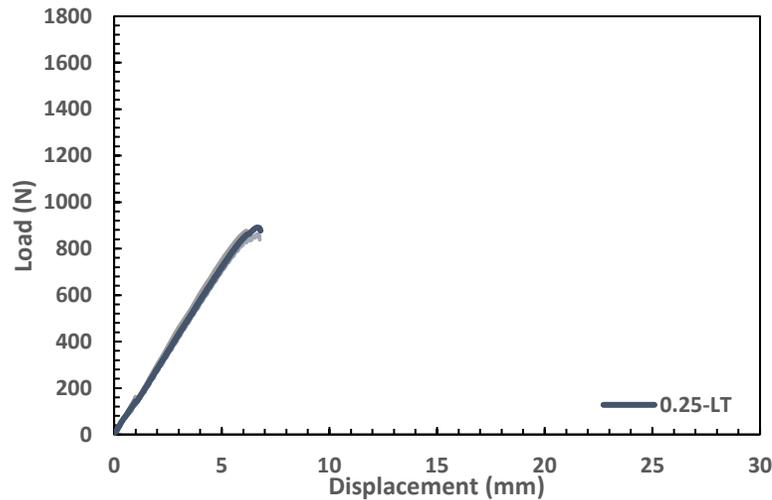
0.475mm



0.625mm



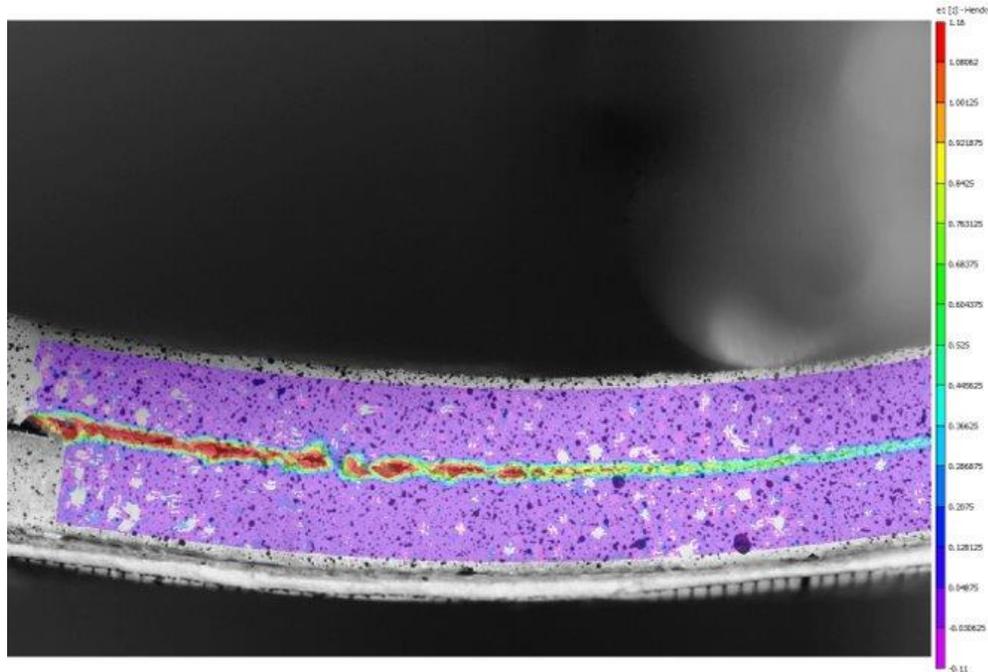
RT



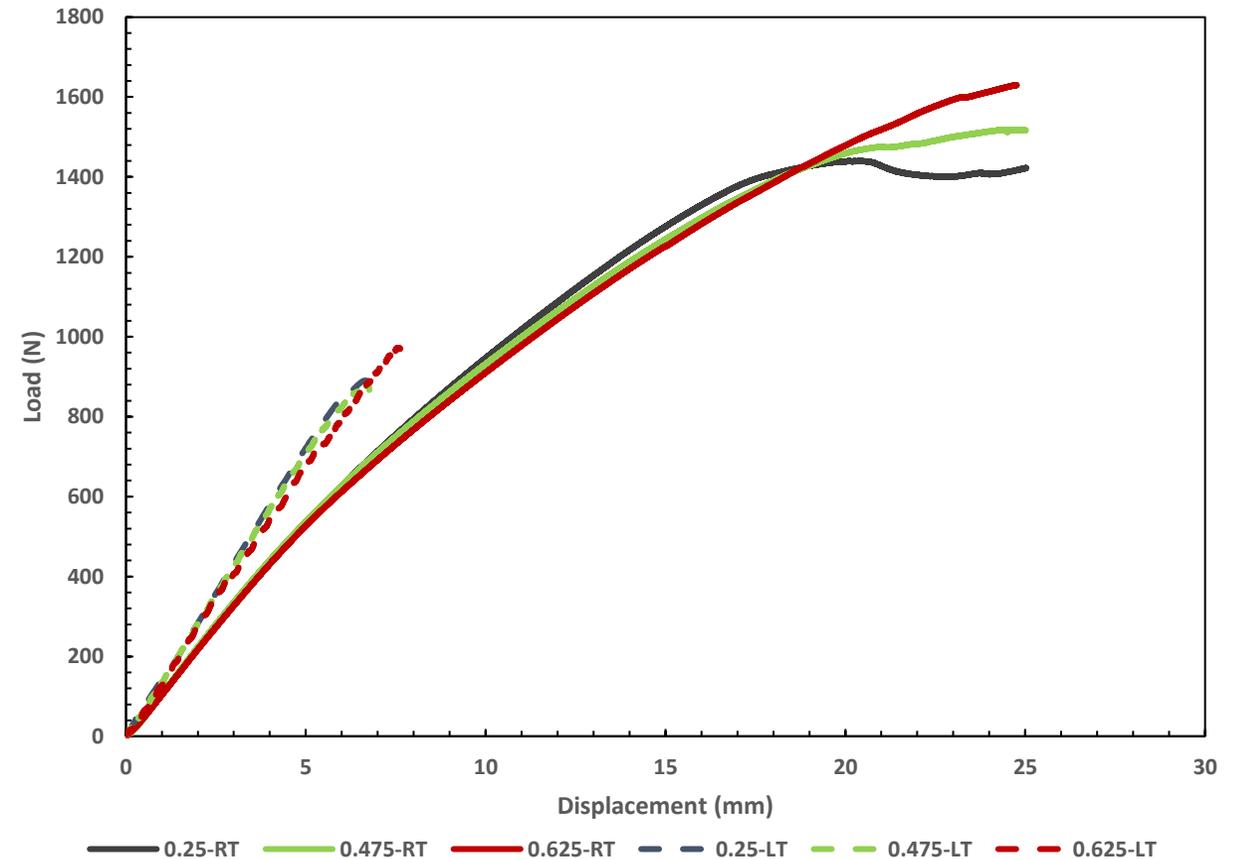
LT

# Experimental Results – Mode-II

- RT and LT  $G_{IIc}$  calculations incomplete due to challenges observing crack-tip during shear deformation
- Trials using DIC were unsuccessful for Mode-II crack tip tracking



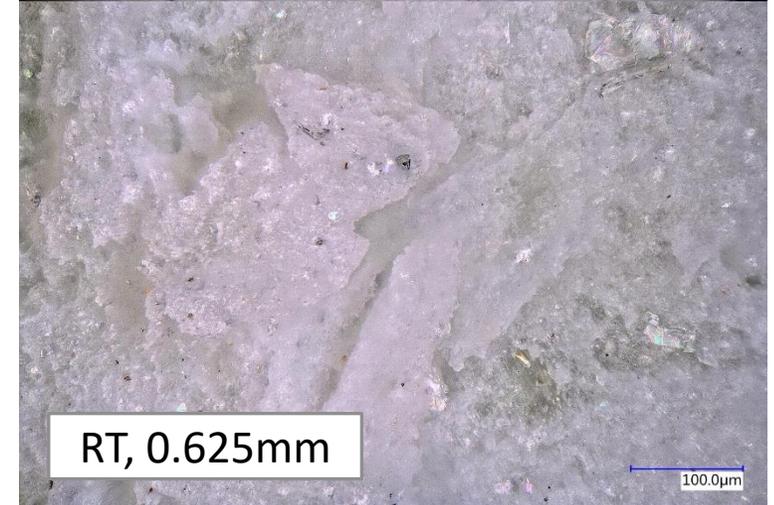
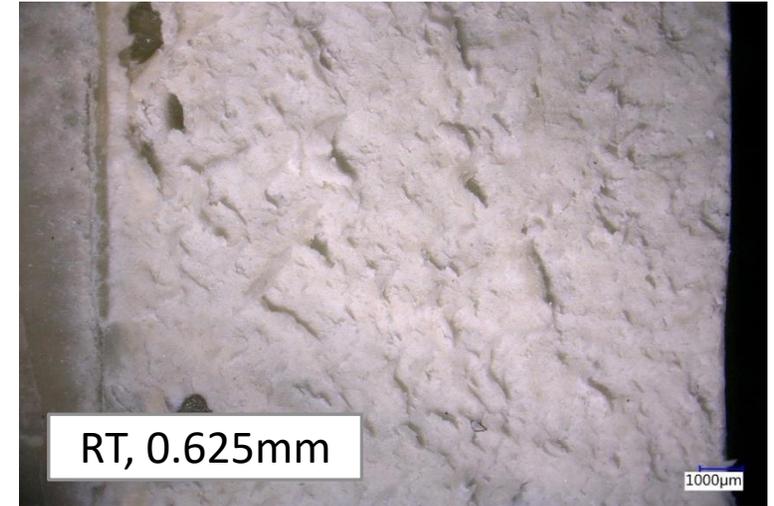
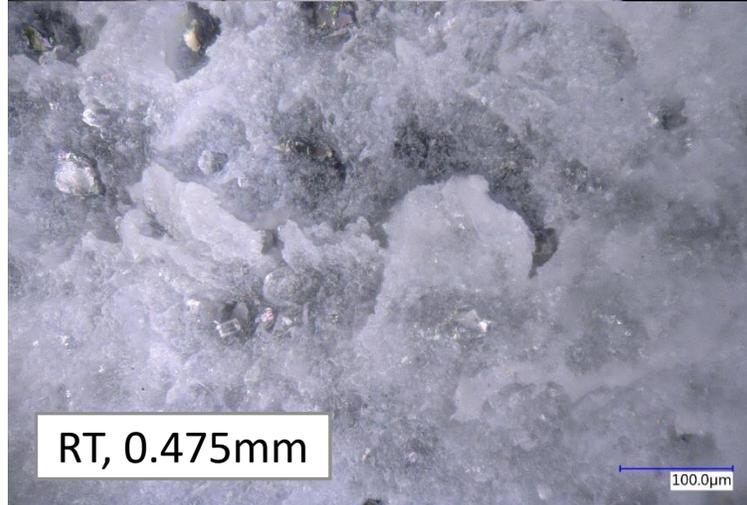
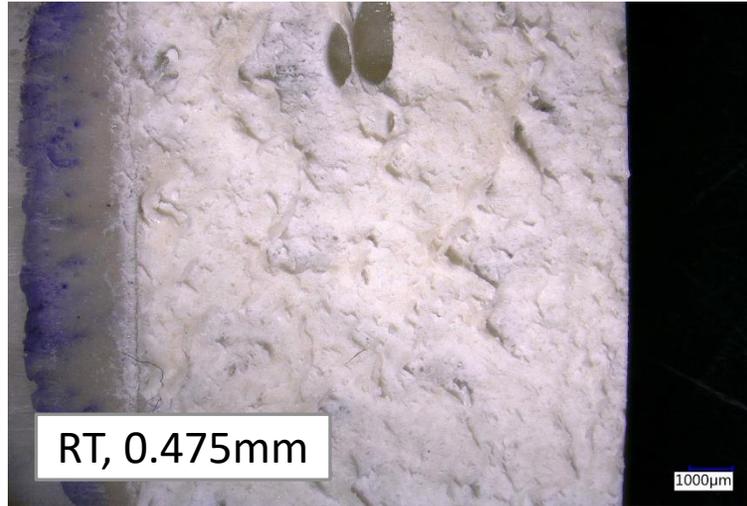
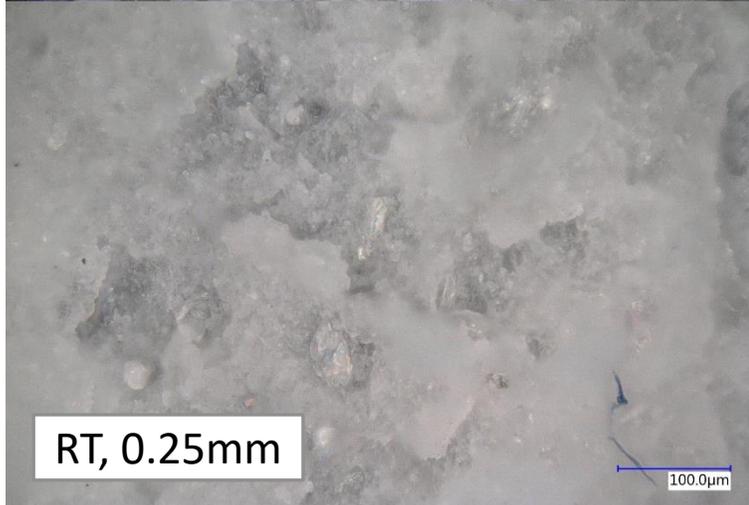
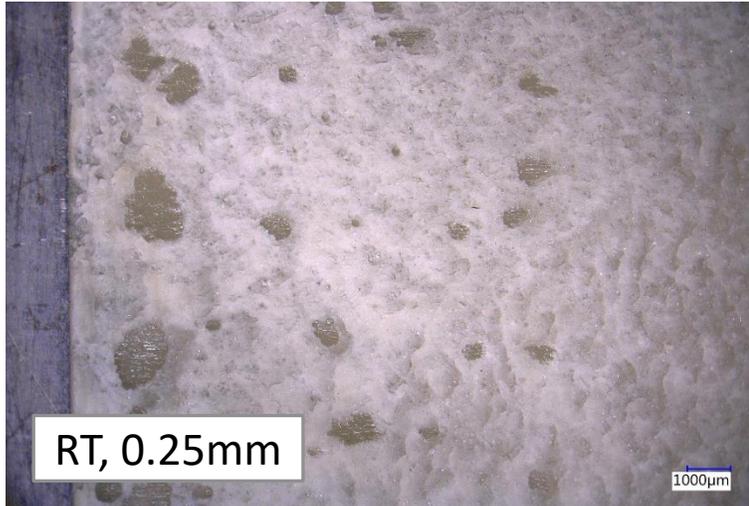
Unsuccessful DIC crack-tip monitoring trial.



Overlay of average Load-Displacement curves for DCB tests at RT and LT.

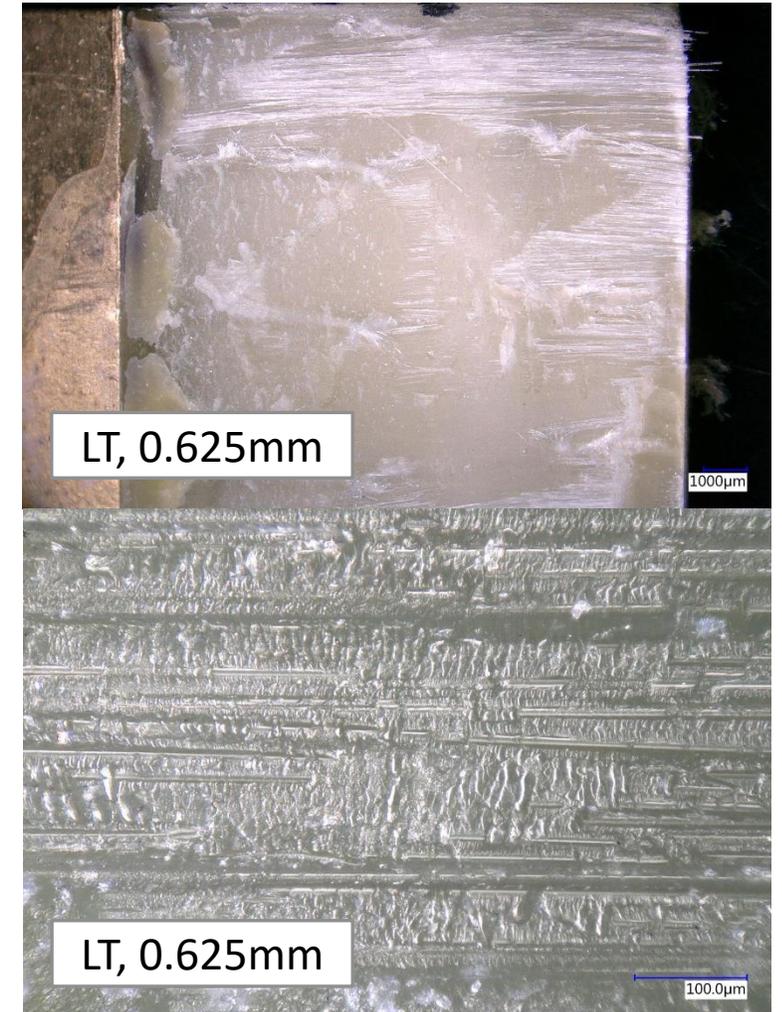
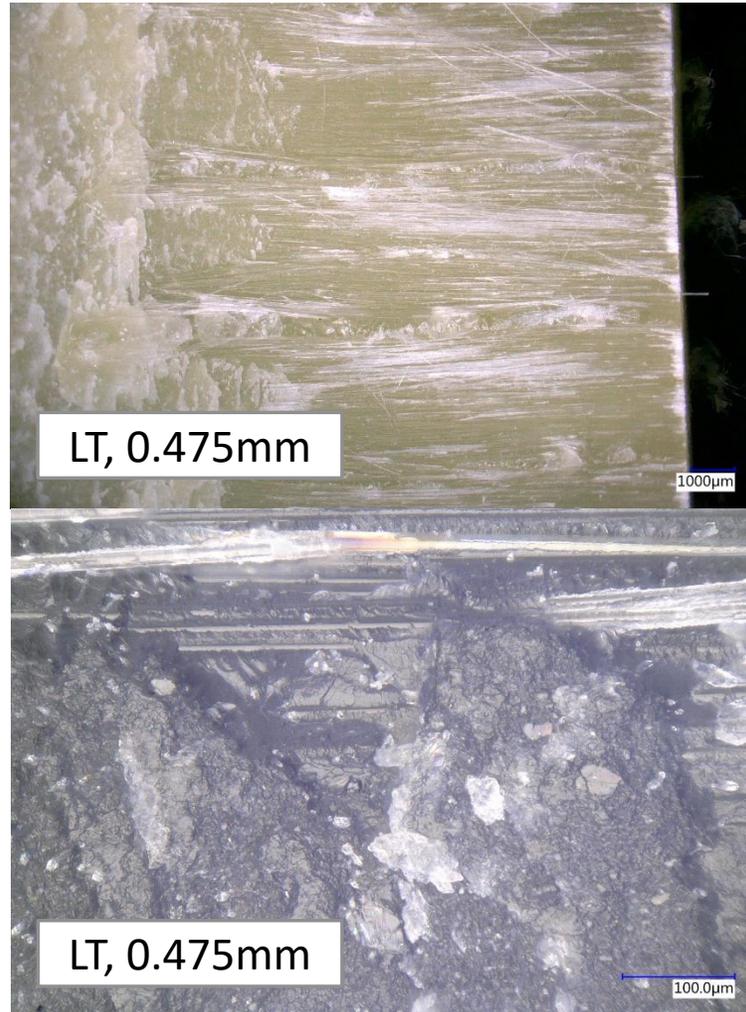
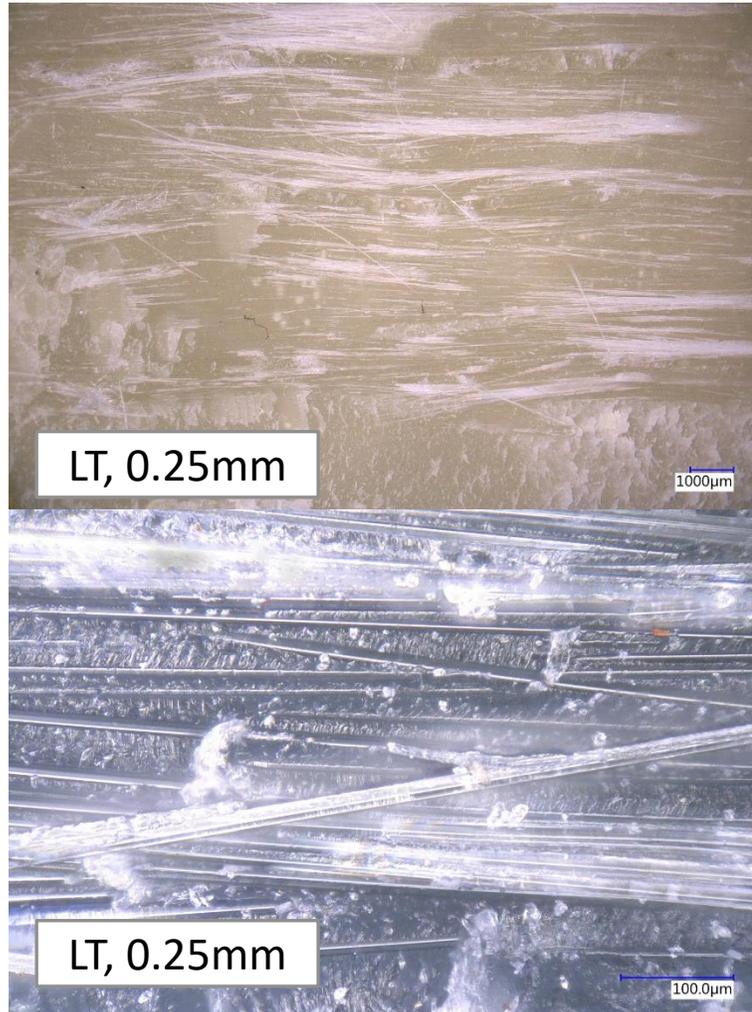
# Fracture Analysis – Mode-II RT

- RT-ENF specimens exhibit hackle formulations indicative of shear failure



# Fracture Analysis – Mode-II LT

- LT-ENF specimens experienced substrate failure (brittle deformation visible under microscope)



# Conclusions and Next Steps

- **Key finding:** at extreme low temperatures (-40°C), the methacrylate adhesive experienced brittle cohesive failure during Mode-I testing and substrate failure during Mode-II testing.
  - **At RT, peak load and displacement at failure increases with increasing bond-line thickness.**
  - **At LT, primary modes of failure are combined brittle cohesive and interfacial failure (for each thickness).**
    - Varying coefficients of thermal expansion may contribute to stress concentrations within substrate (particularly around stitching sites).
- 
- Obtaining  $G_{IC}$  for LT conditions requires performing additional LT-DCB tests.
  - Determine  $G_{IIC}$  for RT and LT conditions using “effective” crack length method.
  - Investigate influence of surface pretreatments on adhesive bond strength (e.g., laser etching and grit-blasting).

# Acknowledgements

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- Special thank you to Devon Hartlen (CRG) for assistance with lab training and specimen bonding.
- Special thank you to Professor Marco Alfano for feedback and input in adhesive bonding theory.



# References

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5. “ASTM D5528/D5528M-21 Standard Test Method for Mode I Interlaminar Fracture Toughness of Unidirectional Fiber-Reinforced Polymer Matrix Composites 1,” ASTM, vol. 15, no. 3, Jan. 2022.
6. B. Chen and D. A. Dillard, “The effect of the T-stress on crack path selection in adhesively bonded joints,” *International Journal of Adhesion and Adhesives*, vol. 21, no. 5, pp. 357–368, 2001.