

# Adaptive Multi-Fidelity & Multi-Scale Modelling of Damage in Composites



**Tong-Earn (T.E.) Tay**

Professor, Dept of Mechanical Engineering,  
National University of Singapore

V.B.C. Tan – colleague

L.H. Poh - colleague

Karthikayen Raju – Postdoc

K.H. Leong – PhD student

K.M. Yeoh – PhD student

Past Students/Postdocs:

Zhi Jie – Tongji University

Boyang Chen – TU Delft

Lu Xin – Tokyo University

## Outline

### Introduction

- Hi-Fidelity Modeling of Progressive Damage in Composites
- Smeared Crack Models (SCM)
- Discrete Crack Models (DCM)
- Separable Cohesive Elements (SCE)

### Numerical Models

- Adaptive Discrete-Smeared Crack Model (A-DiSC)
- Adaptive Multi-Fidelity Model
- Multiscale Modeling with Direct FE<sup>2</sup>

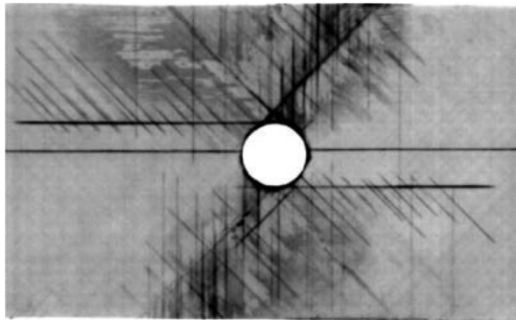
### Applications

- Open-hole Tension
- Static Indentation
- Low-Velocity Impact and Compression After Impact

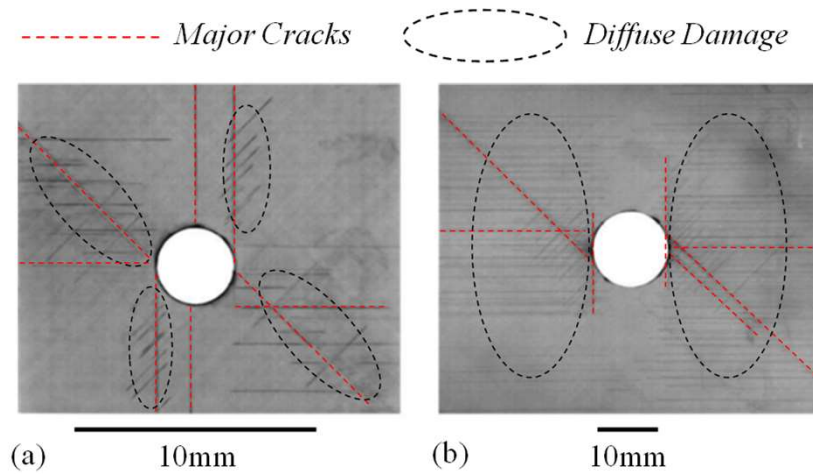
### Outlook & Conclusion

- Some thoughts for future research

## Discrete and Diffused Cracks Co-Exist in Composites



Hallett SR *et al.*, 2009



*Distinct matrix cracks*

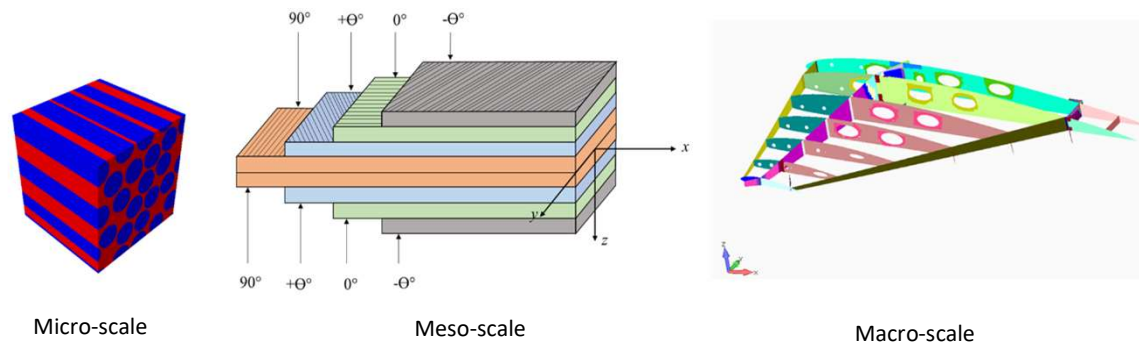
*Diffused microcracks*

*Delaminations*

*Fibre Damage*

\*Hallett SR, Green BG, Jiang WG, Wisnom MR. An experimental and numerical investigation into the damage mechanisms in notched composites. *Composites Part A: Applied Science and Manufacturing*. 2009;40(5):613-24.

## Modeling Challenges



*Interlaminar damage:*  
Cohesive elements

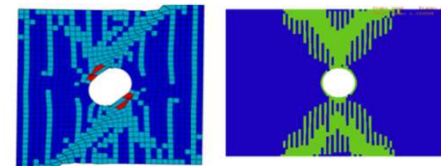
*Intralaminar damage:*  
Cracks  $\rightarrow$  Smeared or Discrete

### Discrete Crack Methods (DCM)

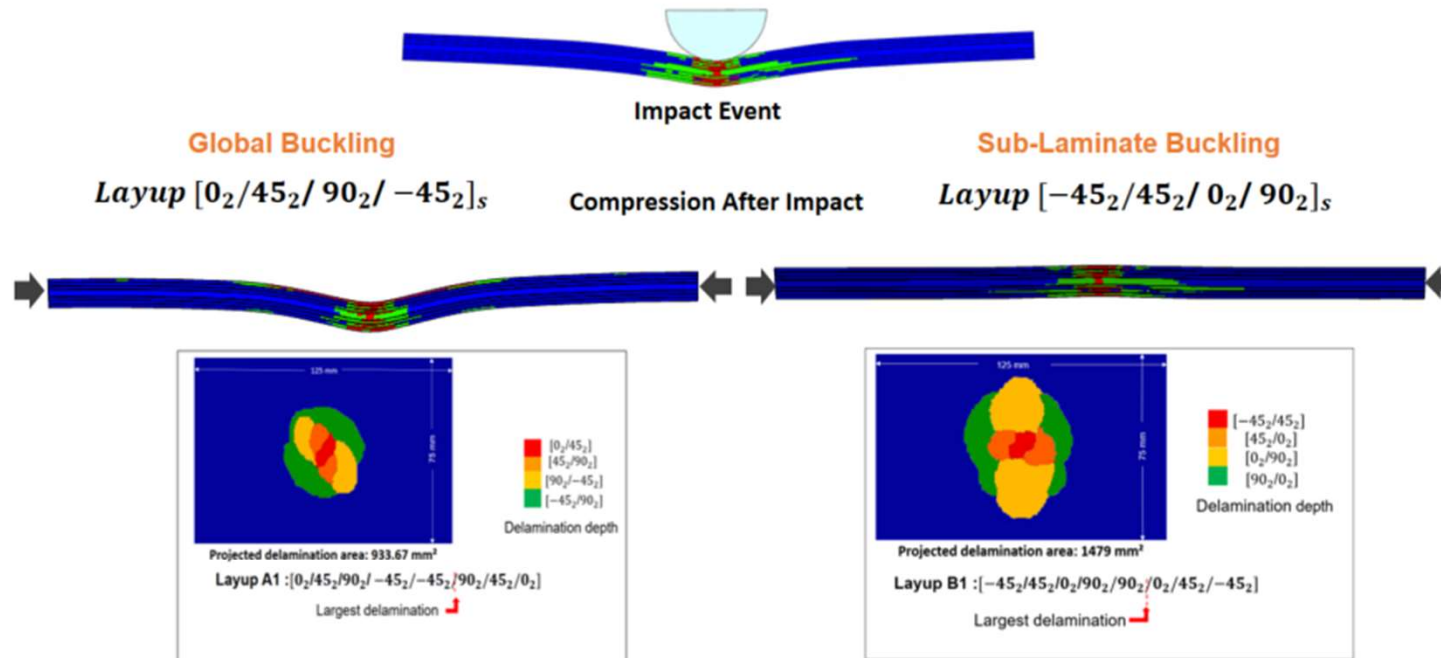
- X-FEM
- Phantom Node Method (PNM)
- Augmented FEM (AFEM)
- Floating Node Method (FNM)

### Smeared Crack Method (SCM)

Damage is smeared out over the element - degradation of the stiffness



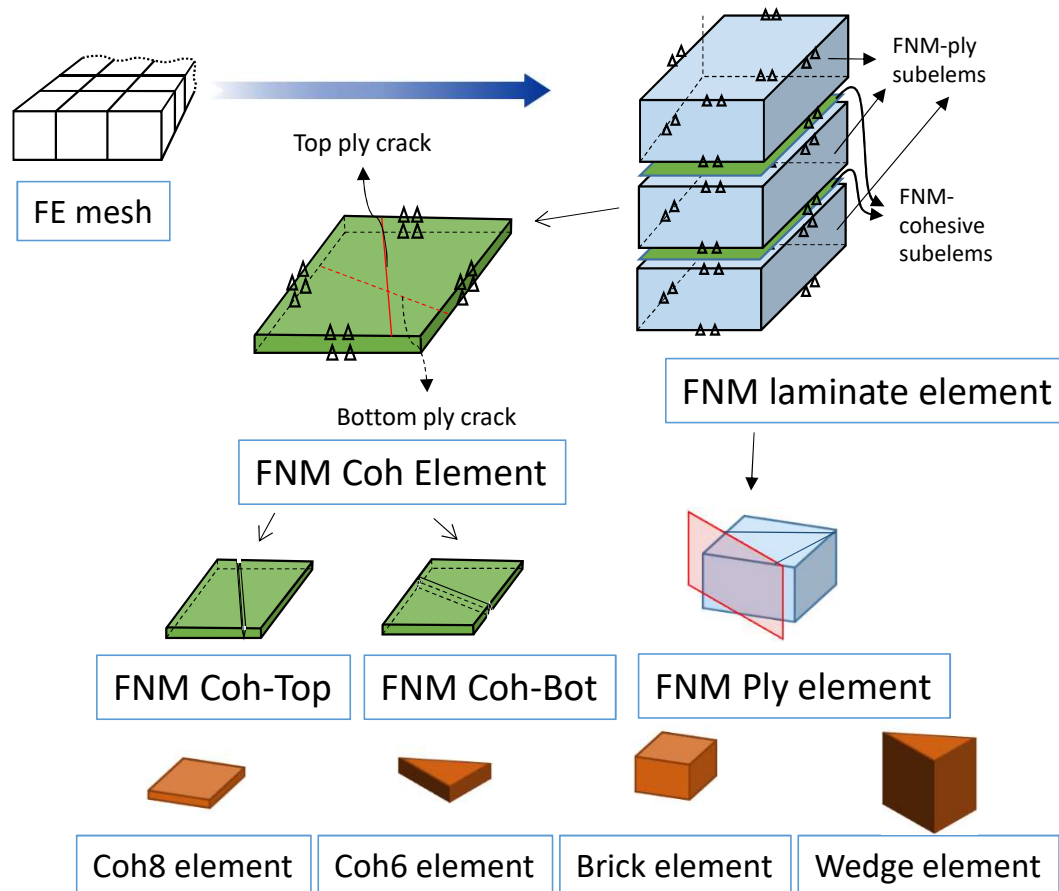
Example of a **Smeared Crack + Cohesive Delamination Model**  
Barely-Visible Impact Damage (BVID) and Compression After Impact (CAI)



- Abir M.R., Tay T.E., Ridha M., Lee H.P., "On the relationship between failure mechanism and compression after impact (CAI) strength in composites", *Composite Structures*, (2017), vol 182, p242-250.
- Abir M.R., Tay T.E., Ridha M., Lee H.P., "Modelling damage growth in composites subjected to impact and compression after impact", *Composite Structures*, (2017), vol 168, p13-25.

## Discrete Crack Model

Example: The Floating Node Method (FNM) and Separable Cohesive Elements (SCE)



Boyang Chen – TU Delft

Lu Xin – Tokyo U

- Lu X., Chen B.Y., Tan V.B.C., Tay T.E., "A separable cohesive element for modelling coupled failure in laminated composite materials", *Composites Part A: Applied Science & Manufacturing*, (2018), vol 107, p387-398.
- Chen B.Y., Pinho S.T., De Carvalho N.V., Baiz P.M., Tay T.E., "A floating node method for the modelling of discontinuities within a finite element", *Engineering Fracture Mechanics*, (2014), vol 127, p104-134.

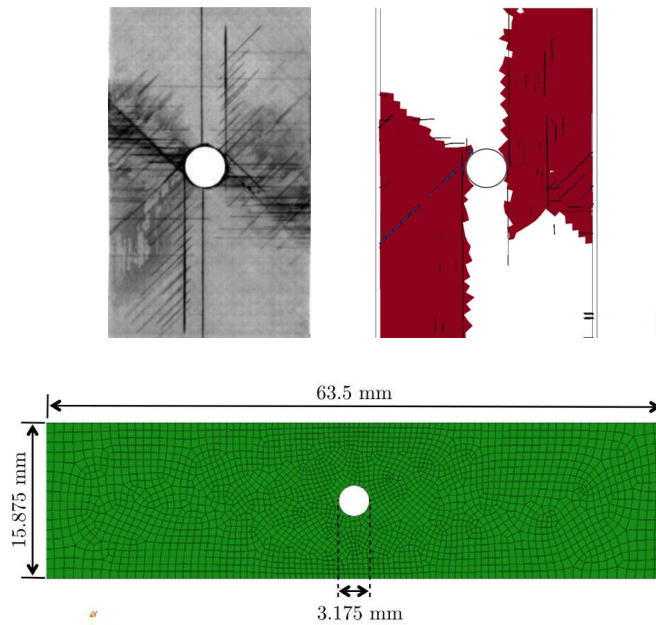


## Discrete Crack Model

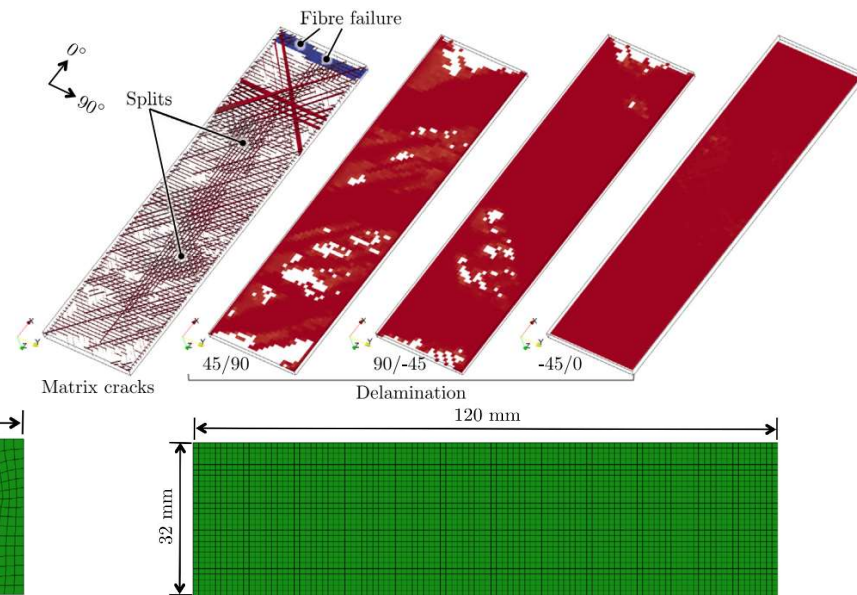
Applied to OHT and unnotched laminates

Boyang Chen – TU Delft

Holed laminate  $[45_4/90_4/-45_4/0_4]_s$



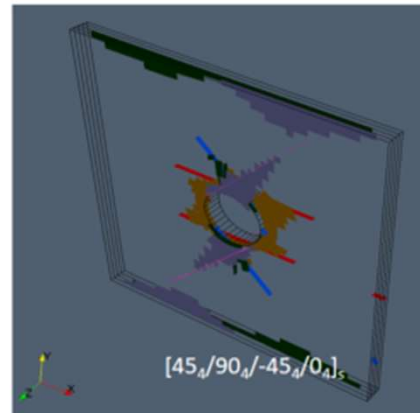
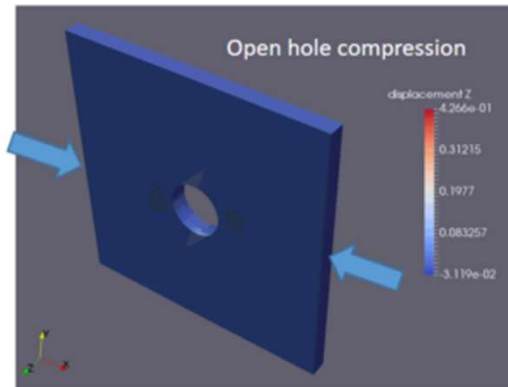
Unnotched laminate  $[45_4/90_4/-45_4/0_4]_s$



- Chen B.Y., Tay T.E., Pinho S.T., Tan V.B.C., "Modelling the tensile failure of composites with the floating node method", *Computer Methods in Applied Mechanics and Engineering*, (2016), vol 308, p414-442.

## Discrete Crack Model

Applied to OHC with geometrical non-linearity

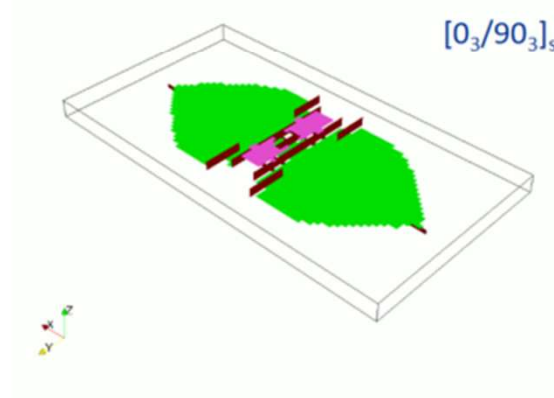


Jie Zhi – Postdoc

- Zhi J., Tay T.E., "Interrogating failure mechanisms of notched composites through a discrete crack modeling approach", *Composites Science & Technology*, (2020), vol 196, 108203.

## Discrete Crack Model

Applied to low velocity impact of composites



- Zhi J., Chen B.Y., Tay T.E., "Geometrically nonlinear analysis of matrix cracking and delamination in composites with floating node method", *Computational Mechanics*, (2019), vol 63, no 2, p201-217.



## Adaptive Discrete-Smeared Crack (A-DiSC)

### Discrete Crack Model (DCM):

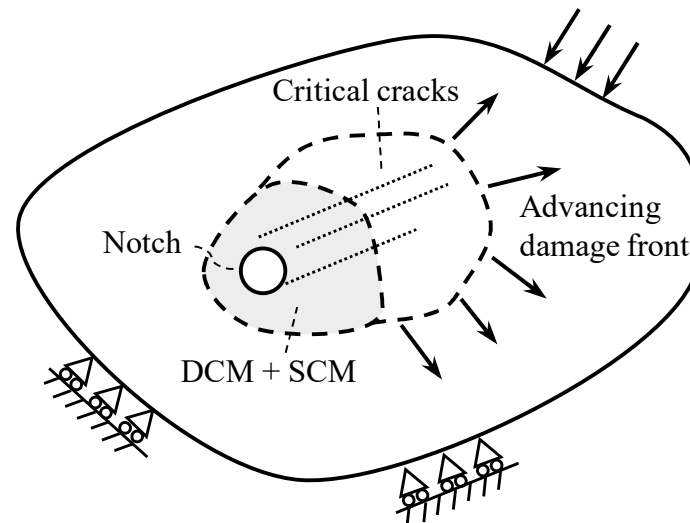
- Closer fidelity to physics of fracture.
- Computationally intense.
- Not practical for larger structures.

### Smeared Crack Model (SCM):

- Efficient representation of numerous micro-cracks.
- May have convergence and distortion issues
- Generally unable to model interaction with delamination accurately.

### A-DiSC / Coupled Model:

- Fidelity to physics of fracture where and when needed, i.e. at active damage fronts.
- Possibly less computationally intense for larger structures.



- Lu X., Ridha M., Tan V.B.C., Tay T.E., "Adaptive discrete-smeared crack (A-DiSC) model for multi-scale progressive damage in composites", *Composites Part A: Applied Science & Manufacturing*, (2019), vol 125, 105513

## Adaptive Discrete-Smeared Crack (A-DiSC)

Lu Xin – Tokyo U

A-DiSC Transition Criterion

Pure DCM



Coupled DCM/SCM

Energy Consistency  
for Model Transition

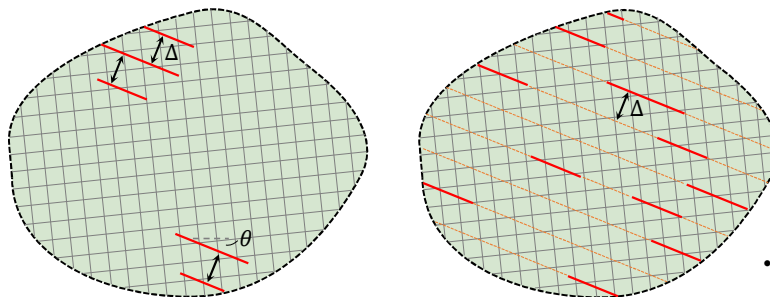
1

*When to transit from DCM to coupled DCM/SCM?*

When matrix cracks are “saturated”

Crack spacing  
 $\Delta = 0.5 \text{ mm}$

— Matrix crack    - - - Crack path



2

*Are there a few larger “critical” cracks which may still grow?*

Characteristic  
length  
 $l_c = 2 \text{ mm}$

- Lu X., Ridha M., Tan V.B.C., Tay T.E., “Adaptive discrete-smeared crack (A-DiSC) model for multi-scale progressive damage in composites”, *Composites Part A: Applied Science & Manufacturing*, (2019), vol 125, 105513

## Adaptive Discrete-Smeared Crack (A-DiSC)

Lu Xin – Tokyo U

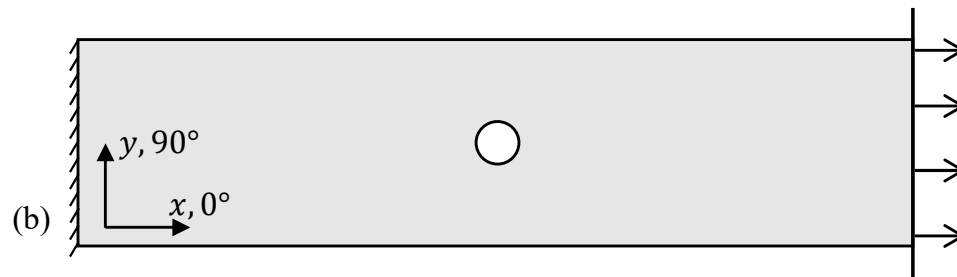
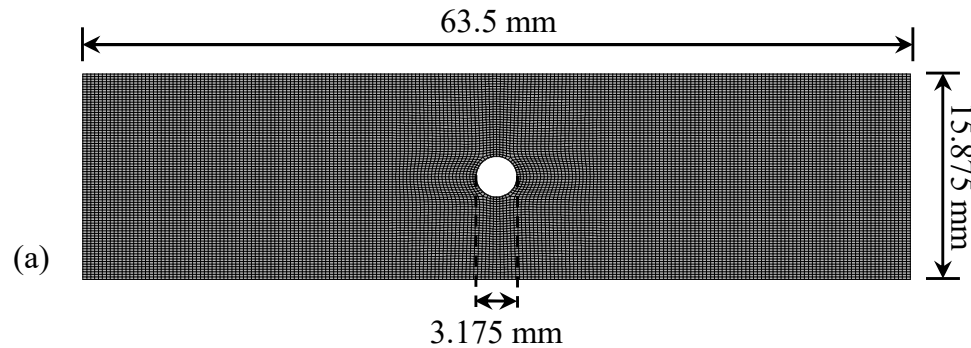
### ❖ Open-hole-tension (OHT)

Layup:  $[45_4/90_4/-45_4/0_4]_s$

Materials: IM7/8552 \*

0.2mm mesh

0.5mm crack spacing



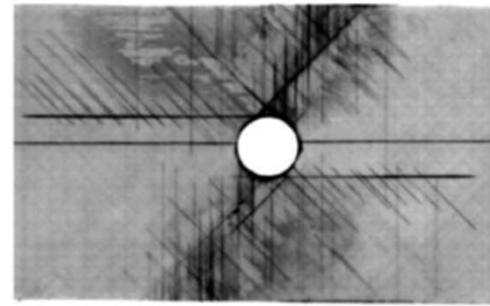
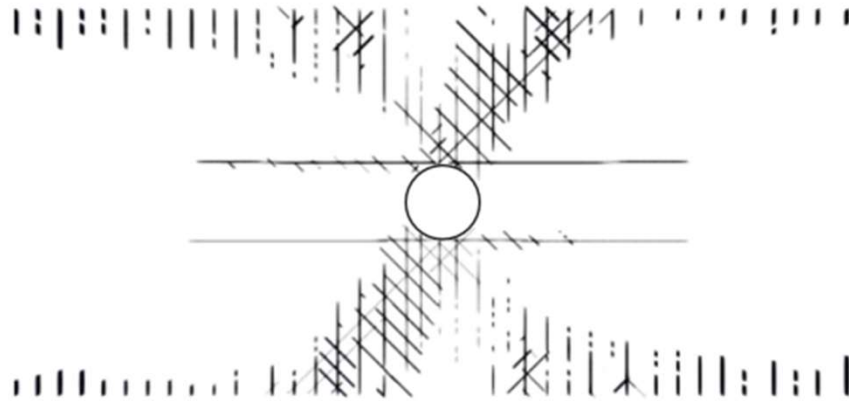
### Models:

1. DCM (benchmark)

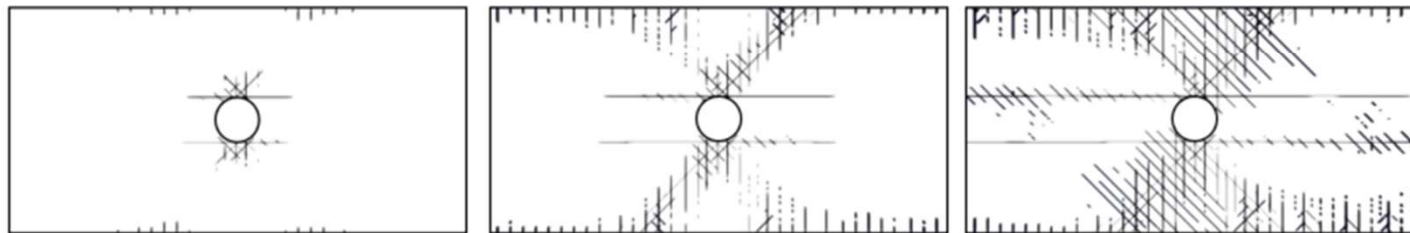
2. A-DiSC

\*B.G. Green, M.R. Wisnom, S.R. Hallett, An experimental investigation into the tensile strength scaling of notched composites, *Compos. Part A Appl. Sci. Manuf.*, 38 (2007) 867-878.

## ❖ Pure Discrete Crack Model

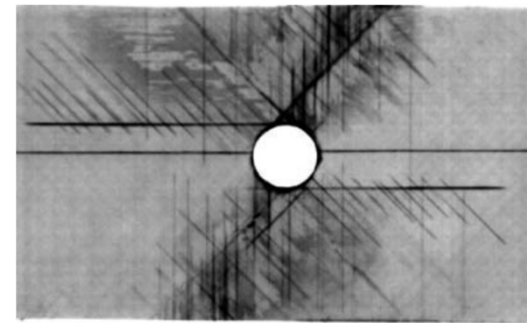
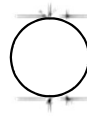


S.R. Hallett *et al.*, 2009

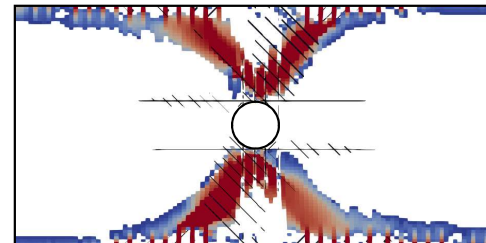
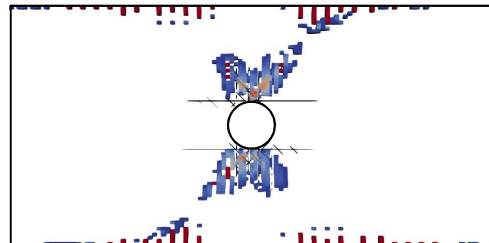


## ❖ Adaptive Discrete-Smeared Crack Model

90° ply

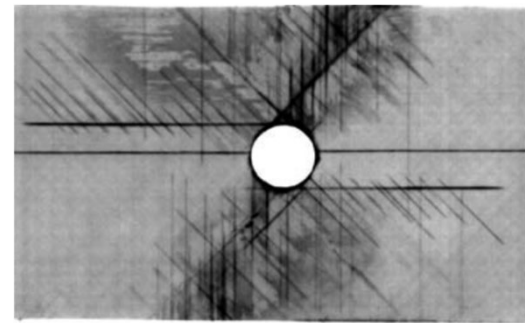
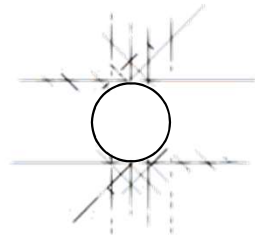


S.R. Hallett *et al.*, 2009

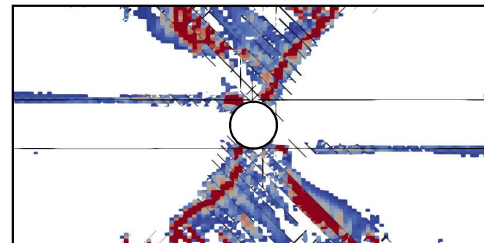
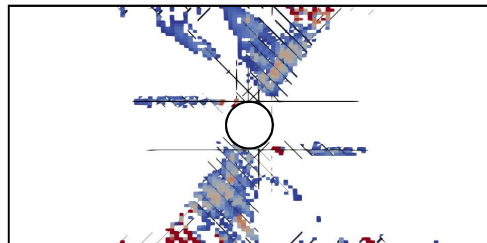


## ❖ Adaptive Discrete-Smeared Crack Model

-45° ply (internal)



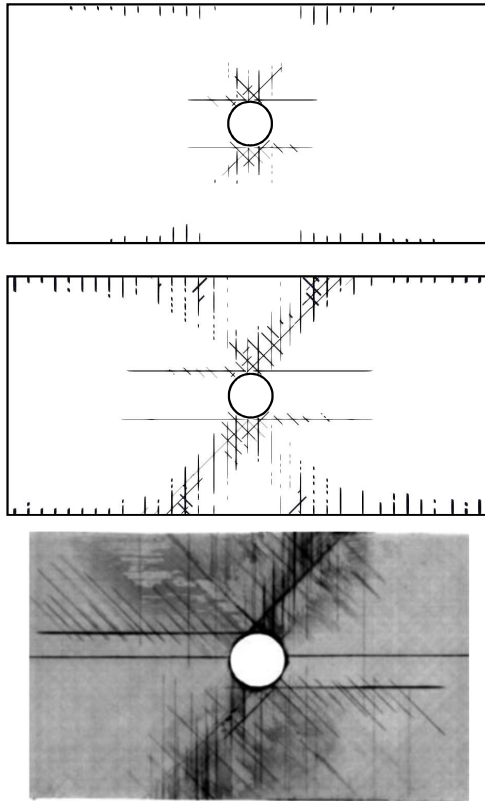
S.R. Hallett *et al.*, 2009



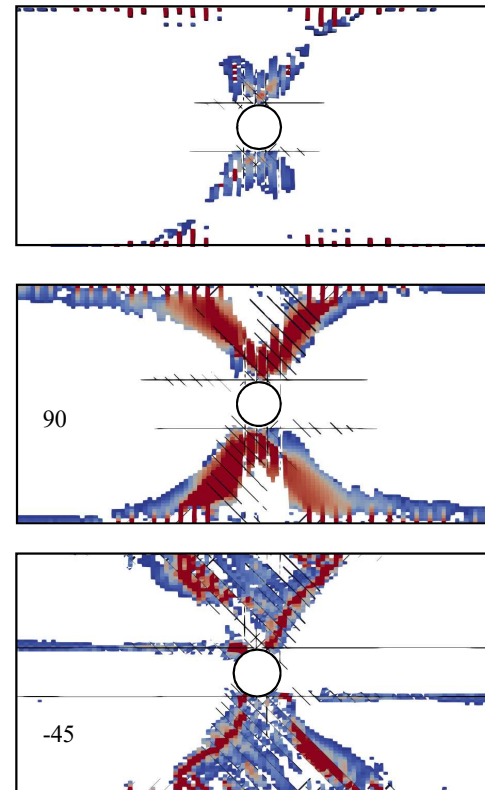


## ❖ Comparison

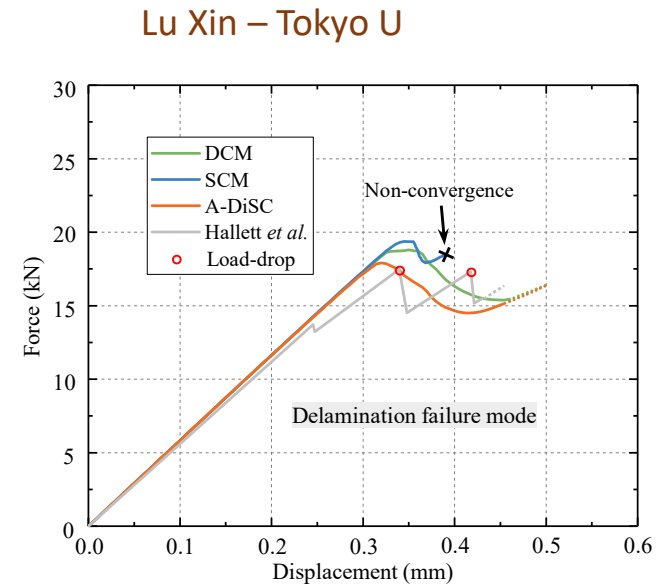
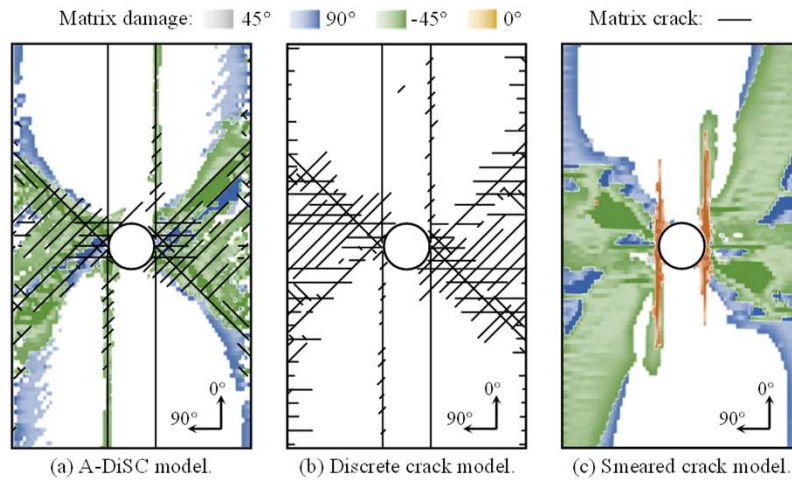
### Full 3D Discrete Crack Model



### Adaptive Discrete-Smeared Crack Model



## ❖ Comparison



Models	DoFs	Iterations	CPU time (s)
SCM	412704	N.A.	N.A.
DCM	705012	4089	1066660
A-DiSC ( $\rho = \rho_s, l_c = 2 \text{ mm}$ )	538839	3681	966364
A-DiSC ( $\rho = 0.3 \rho_s, l_c = 2 \text{ mm}$ )	446202	3726	825149

$\rho_s$ : saturated matrix crack density

Up to 36.7% saving

Up to 22.6% saving

- Lu X., Ridha M., Tan V.B.C., Tay T.E., "Adaptive discrete-smeared crack (A-DiSC) model for multi-scale progressive damage in composites", *Composites Part A: Applied Science & Manufacturing*, (2019), vol 125, 105513

## Adaptive Multi-Fidelity (AMF) Model

### Discrete Crack Model (DCM):

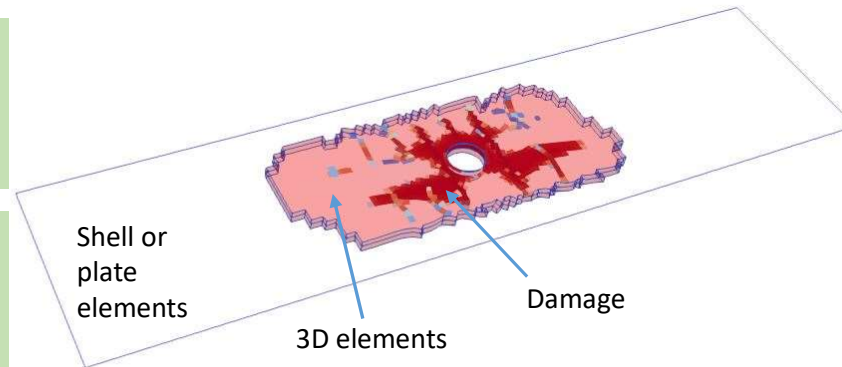
- Closer fidelity to physics of fracture.
- Computationally intense.
- Not practical for larger structures.

### Smeared Crack Model (SCM):

- Efficient representation of numerous micro-cracks.
- May have convergence and distortion issues
- Generally unable to model interaction with delamination accurately.

### A-DiSC / Coupled Model:

- Fidelity to physics of fracture where and when needed, i.e. at active damage fronts.
- Possibly less computationally intense for larger structures.



### Adaptive Multi-Fidelity (AMF) Model:

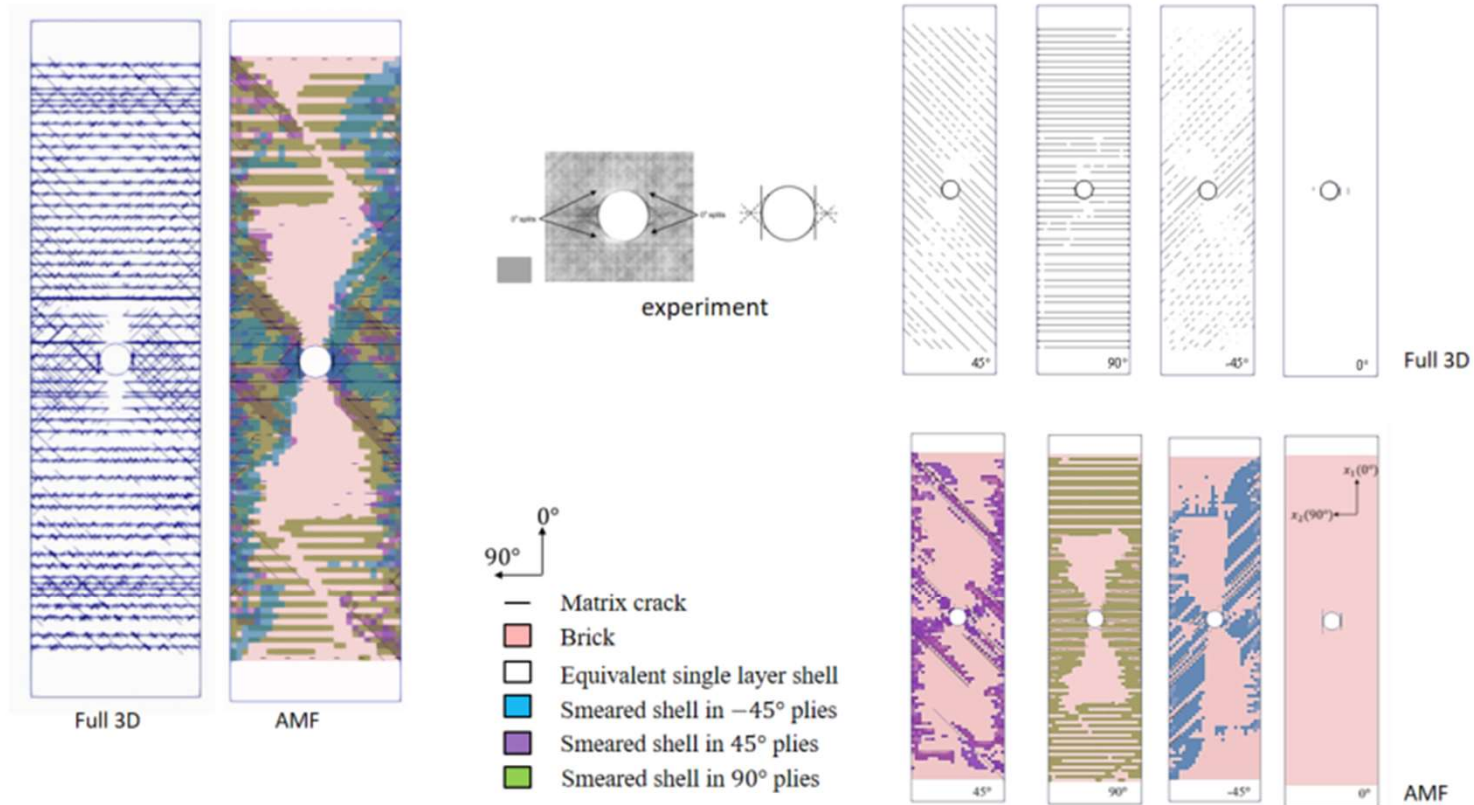
- Fidelity to physics of fracture where and when needed, i.e. at active damage fronts.
- **Shell→3D→Shell elements adaptivity**
- Less computationally intense for larger structures.

Leong K.H., Zhi J., Lee H.P., Tan V.B.C., Tay T.E., "Adaptive multi-fidelity (AMF) modelling of progressive damage in notched composite laminates", Composites Part A: Applied Science & Manufacturing, <https://doi.org/10.1016/j.compositesa.2021.106790> (2022), vol 154, 106790.

## Adaptive Multi-Fidelity (AMF) Model

K.H. Leong – PhD student

Open hole tension (OHT)  $[45/90/-45/0]_{4s}$

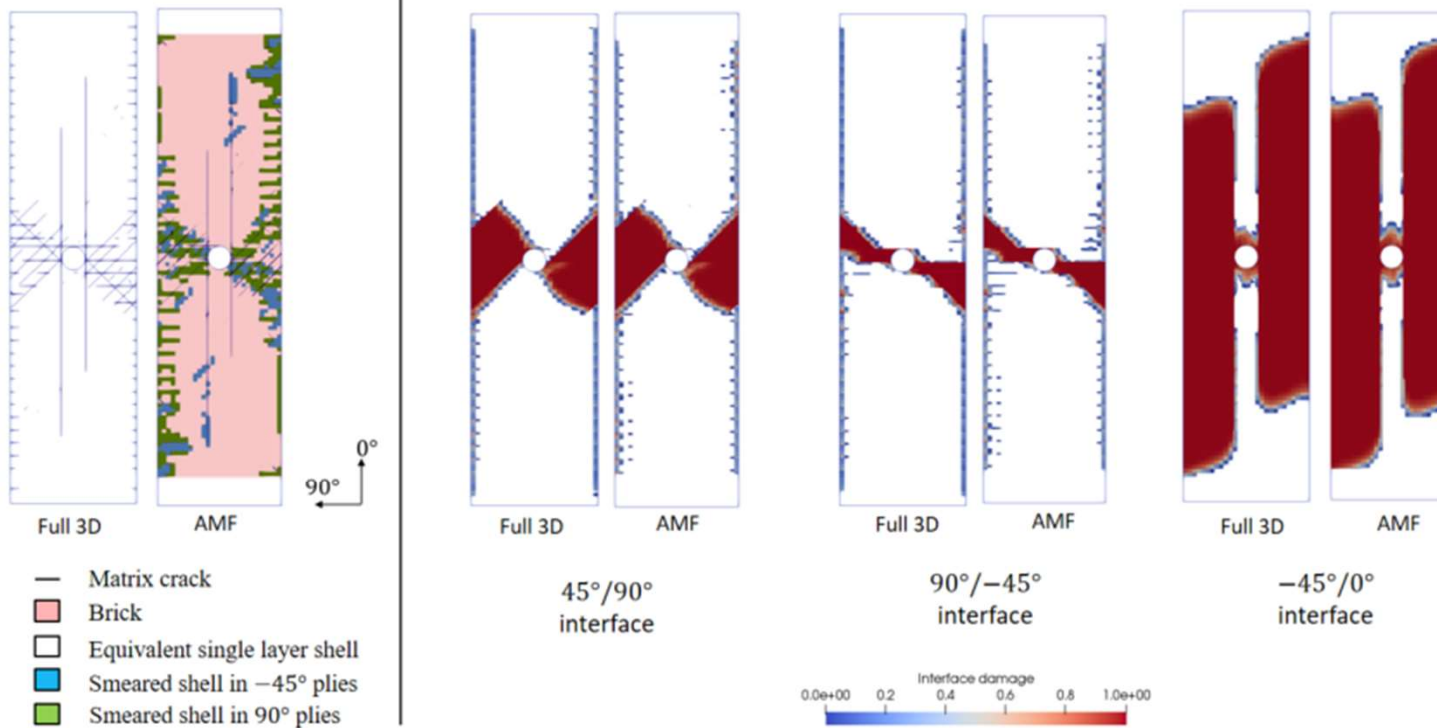


Leong K.H., Zhi J., Lee H.P., Tan V.B.C., Tay T.E., "Adaptive multi-fidelity (AMF) modelling of progressive damage in notched composite laminates", Composites Part A: Applied Science & Manufacturing, (2022), vol 154, 106790.

## Adaptive Multi-Fidelity (AMF) Model

Open hole tension (OHT)

Layup  $[45_4/90_4/-45_4/0_4]_s$



## Adaptive Multi-Fidelity (AMF) Model

Computational time

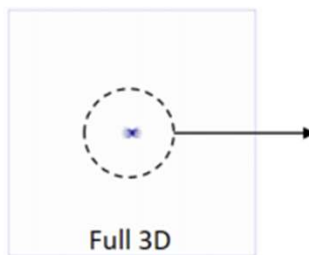
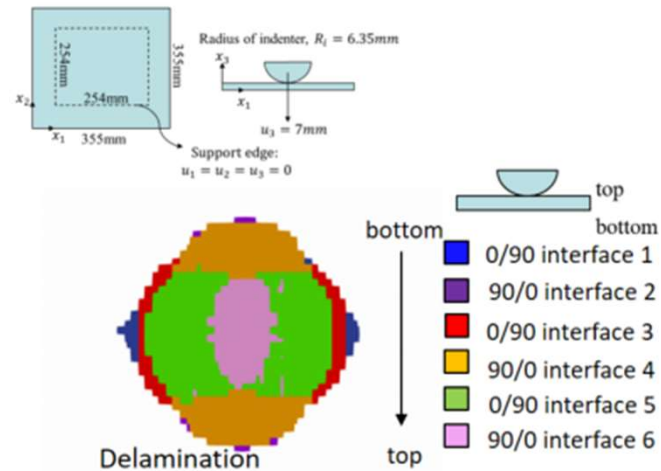
8 CPUs (Xeon(R) CPU E5-2697 @ 2.30GHz)

OHT $[45_4/90_4/-45_4/0_4]_s$			
Total CPU time (hours)	10.5	8.7	➡ -17.1%
Wallclock time (hours)	11.6	9.6	➡ -16.5%
OHT $[45/90/-45/0]_{4s}$			
Total CPU time (hours)	330.8	245.5	➡ -25.7%
Wallclock time (hours)	333.5	252.7	➡ -23.2%



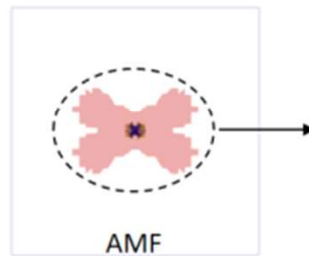
## Adaptive Multi-Fidelity (AMF) Model

Static indentation test  
Cross-ply layup  $[0_2/90_2/0_2/90_2]_s$



Matrix crack

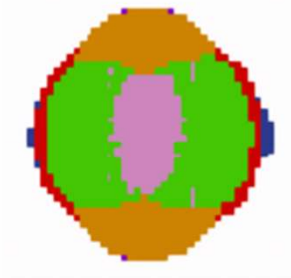
Delamination



- ESL shell elements
- brick elements



- matrix crack
- smeared shell elements (ply 1 (0°))
- smeared shell elements (ply 2 (90°))
- smeared shell elements (ply 3 (0°))



## 22

## Adaptive Multi-Fidelity (AMF) Model

Computational time

-8 CPUs (Xeon(R) CPU E5-2697 @ 2.30GHz)

Static indentation $[0_2/90_2/0_2/90_2]_s$			
	Full 3D	AMF	
Total CPU time (hours)	71.6	30.5	→ -57.4%
Wallclock time (hours)	73.3	33.1	→ -54.7%

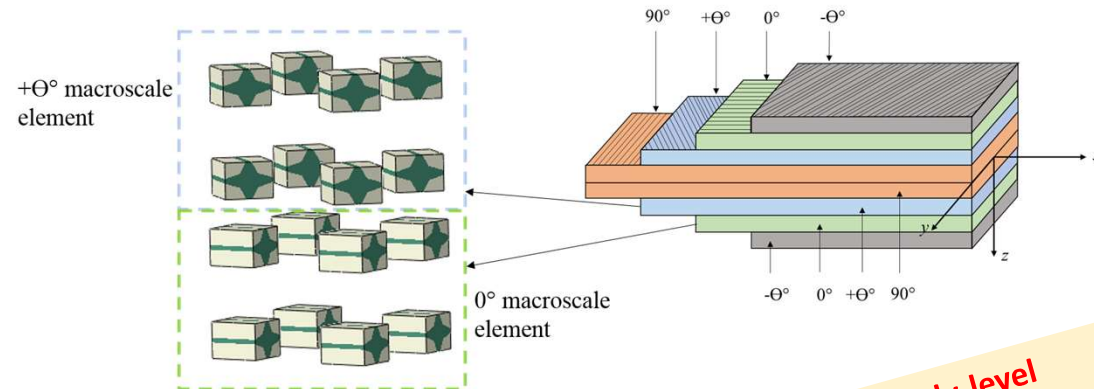
## Related Publications (partial list)

- ✓ Abir M.R., Tay T.E., Ridha M., Lee H.P., “On the relationship between failure mechanism and compression after impact (CAI) strength in composites”, **Composite Structures**, (2017), vol 182, p242-250.
- ✓ Abir M.R., Tay T.E., Ridha M., Lee H.P., “Modelling damage growth in composites subjected to impact and compression after impact”, **Composite Structures**, (2017), vol 168, p13-25.
- ✓ Lu X., Chen B.Y., Tan V.B.C., Tay T.E., “A separable cohesive element for modelling coupled failure in laminated composite materials”, **Composites Part A: Applied Science & Manufacturing**, (2018), vol 107, p387-398.
- ✓ Chen B.Y., Pinho S.T., De Carvalho N.V., Baiz P.M., Tay T.E., “A floating node method for the modelling of discontinuities within a finite element”, **Engineering Fracture Mechanics**, (2014), vol 127, p104–134.
- ✓ Chen B.Y., Tay T.E., Pinho S.T., Tan V.B.C., “Modelling the tensile failure of composites with the floating node method”, **Computer Methods in Applied Mechanics and Engineering**, (2016), vol 308, p414-442.
- ✓ Zhi J., Tay T.E., “Interrogating failure mechanisms of notched composites through a discrete crack modeling approach”, **Composites Science & Technology**, (2020), vol 196, 108203.
- ✓ Zhi J., Chen B.Y., Tay T.E., “Geometrically nonlinear analysis of matrix cracking and delamination in composites with floating node method”, **Computational Mechanics**, (2019), vol 63, no 2, p201-217.
- ✓ Lu X., Ridha M., Tan V.B.C., Tay T.E., “Adaptive discrete-smeared crack (A-DISC) model for multi-scale progressive damage in composites”, **Composites Part A: Applied Science & Manufacturing**, (2019), vol 125, 105513.
- ✓ Lu X., Guo X.M., Tan V.B.C., Tay T.E., “From Diffuse Damage to Discrete Crack: A Coupled Failure Model for Multi-stage Progressive Damage of Composites”, **Computer Methods in Applied Mechanics and Engineering**, (2021), vol 379, 113760.
- ✓ Leong K.H., Zhi J., Lee H.P., Tan V.B.C., Tay T.E., “Adaptive multi-fidelity (AMF) modelling of progressive damage in notched composite laminates”, **Composites Part A: Applied Science & Manufacturing**, (2022), vol 154, 106790.
- ✓ Leong K.H., Zhi J., Lee H.P., Tan V.B.C., Tay T.E., “Adaptive Multi-Fidelity (AMF) modelling of delamination migration under bending”, **Composite Structures**, (2023), vol 305, 116549.



## Concurrent Multiscale Modeling of Composites with Direct FE<sup>2</sup>

Vincent B.C. Tan – colleague



**No need for ply-level failure criteria?**

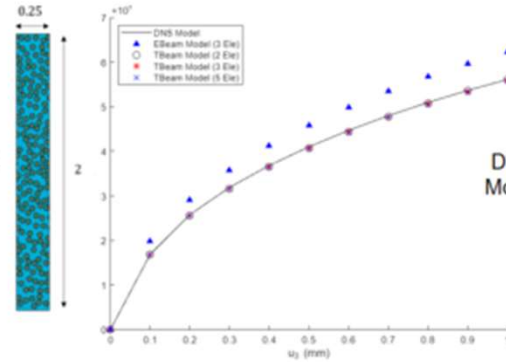
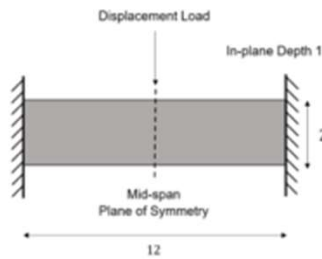
- FE<sup>2</sup> simulations allows for the analyses of structures made from heterogeneous materials without the need for homogenized material models
- Direct FE<sup>2</sup> reduces the two separate and concurrent simulations at macro- and micro-scales required in FE<sup>2</sup> to a single simulation
- Direct FE<sup>2</sup> can be carried out directly on commercial FE codes
- Features already on commercial codes (e.g., finite deformation, viscoplasticity, damage models) remain available without any user intervention

- Tan V.B.C., Raju K., Lee H.P., "Direct FE<sup>2</sup> for concurrent multi level modelling of heterogeneous structures", *Computer Methods in Applied Mechanics and Engineering*, (2019), 112694.
- Raju K., Tay T.E., Tan V.B.C., "A review of the FE<sup>2</sup> method for composites", *Multiscale and Multidiscip. Model. Exp. and Des.*, (2021), vol 4, 1-24.

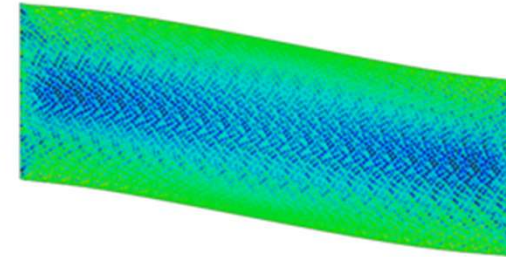
# Multiscale FE<sup>2</sup> modelling of composites using shear-flexible beam elements

K.M. Yeoh – PhD student

All dimensions in mm

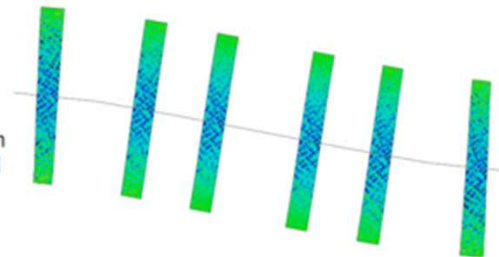


DNS Model

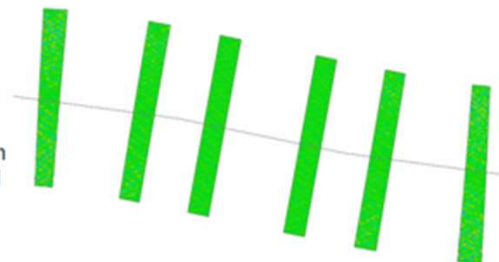


Model	CPU Time (s)	Reduction (%)
DNS	863.8	-
TBeam (2 beam ele)	113.9	86.8
TBeam (3 beam ele)	222.8	74.2
TBeam (5 beam ele)	332.8	61.5

TBeam Model

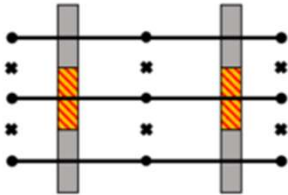


EBeam Model



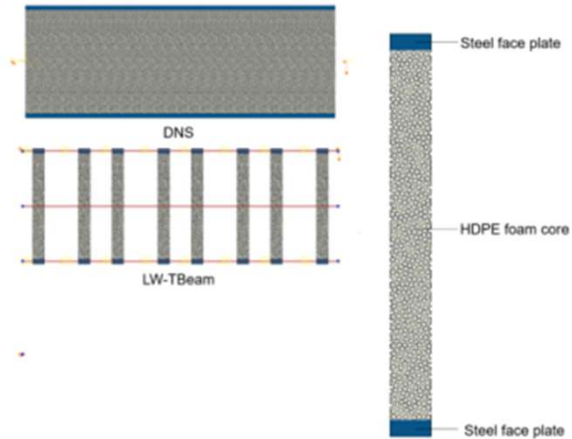


## Foam core sandwich beams with steel facesheets



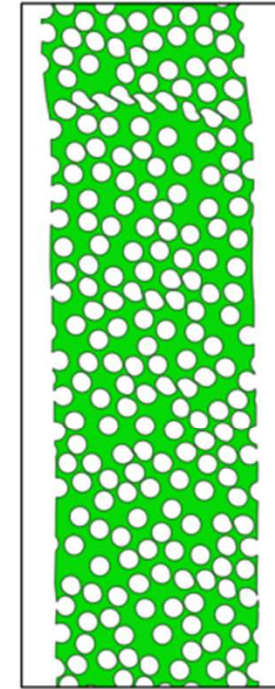
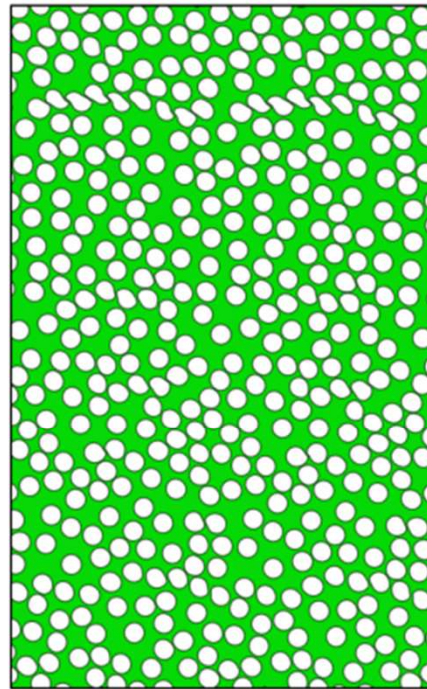
- Each layer is modelled with an individual macroscale beam element
- Modified shear angle constraints to account for layerwise nature
- Ensure displacement continuity between the different layers at the interface

K.M. Yeoh – PhD student



DNS

LW-TBeam

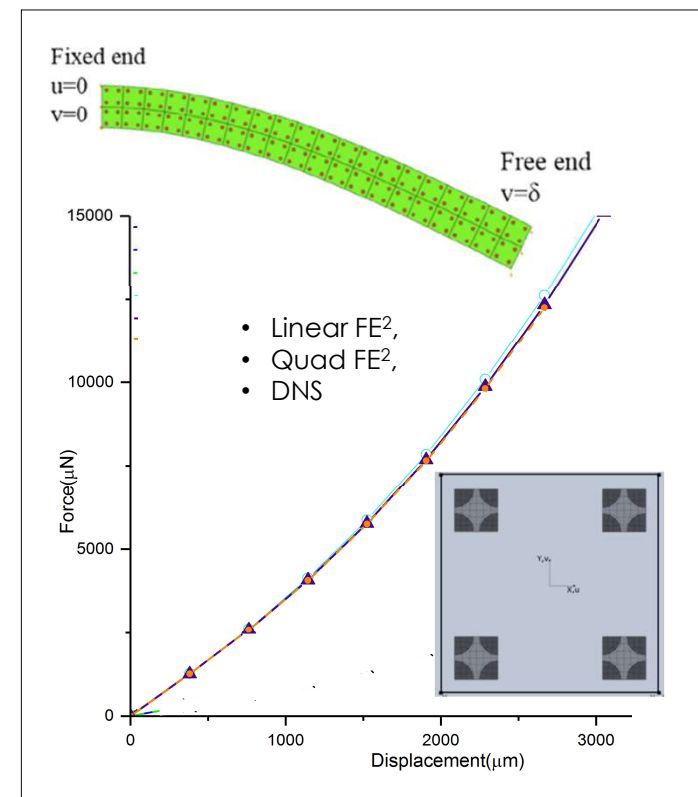


# Direct FE<sup>2</sup> research

Slide courtesy of Vincent Tan

1. **Direct FE<sup>2</sup> for concurrent multilevel modelling of heterogeneous structures, (2020) CMAME, 360, 112694.**
2. A review of the FE<sup>2</sup> method for composites, (2021) Multiscale and Multidisciplinary Modeling, Exp and Des, 4 (1).
3. Direct FE<sup>2</sup> for simulating strain-rate dependent compressive failure of cylindrical CFRP, (2021) Comp Part C, 5, 100165.
4. Analysis of nonlinear shear and damage behaviour of angle-ply laminates with Direct FE<sup>2</sup>, (2021) CST, 216, 109050.
5. Transient multi-scale analysis with micro-inertia effects using Direct FE<sup>2</sup> method, (2021) Comp Mech, 67 (6).
6. Multiscale analysis of thermal problems in heterogeneous materials with Direct FE<sup>2</sup> method, (2021) IJNME, 122 (24).
7. Multiscale computational homogenisation of shear-flexible beam elements: a Direct FE<sup>2</sup> approach, (2022) Comp Mech, 70 (5).
8. Direct FE<sup>2</sup> for concurrent multilevel modelling modeling of heterogeneous thin plate structures, (2022) CMAME, 392, 114658.
9. Direct FE<sup>2</sup> modeling of heterogeneous materials with a micromorphic computational homogenization framework, (2022) CMAME, 393, 114837.

First report of Direct FE<sup>2</sup> using CFRP beam as example. Demonstrated large deformation and material nonlinearity.

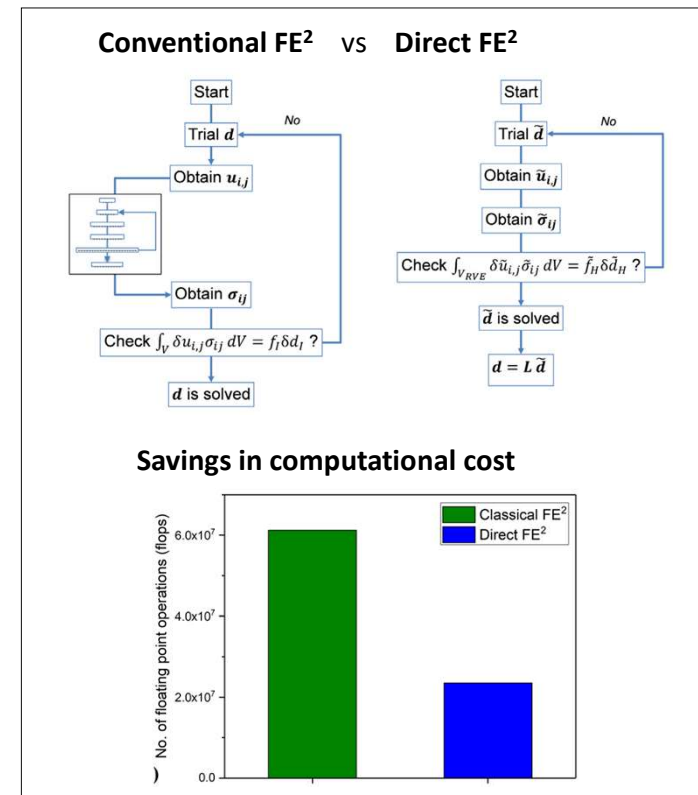


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Reviewed applications of FE<sup>2</sup> in composites and showed how the monolithic scheme of Direct FE<sup>2</sup> reduces computational cost.



## Direct FE<sup>2</sup> research

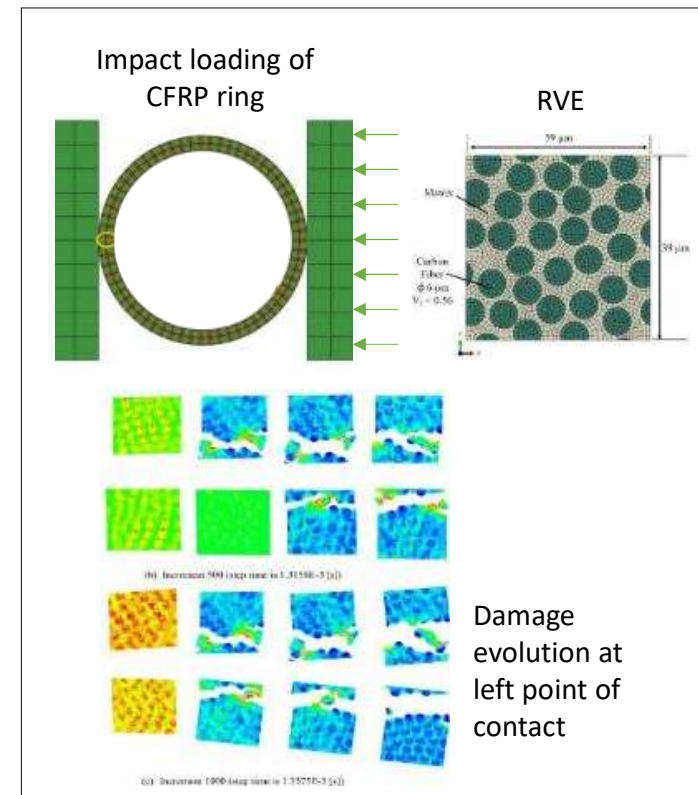
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In Collaboration with Jun Koyanagi,  
Tokyo University of Science.



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Using substantially larger RVEs, evolution of microstructural damage in rate-sensitive carbon-epoxy composite was modeled.

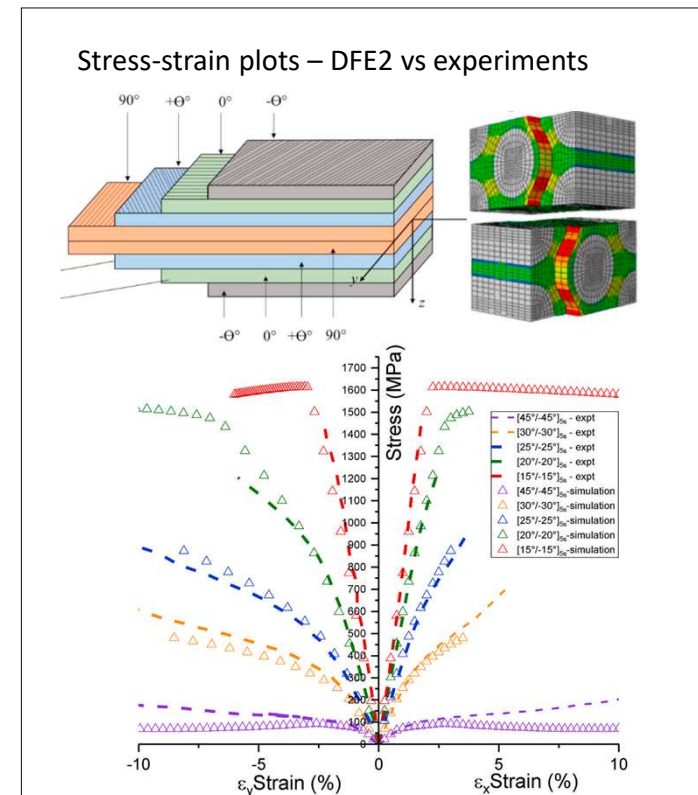


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DFE<sup>2</sup> enabled interaction of fibre rotation, matrix plasticity and cracks to predict  $\sigma$  vs  $\epsilon$  plots of angle-ply matching experiments

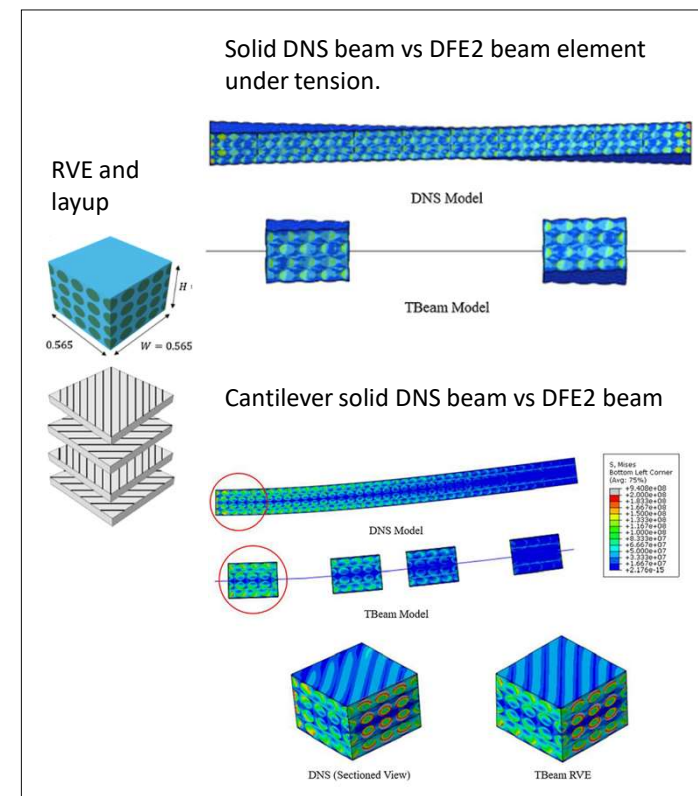


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DFE<sup>2</sup> has been extended to shear deformable beam elements with examples on CFRP beams.



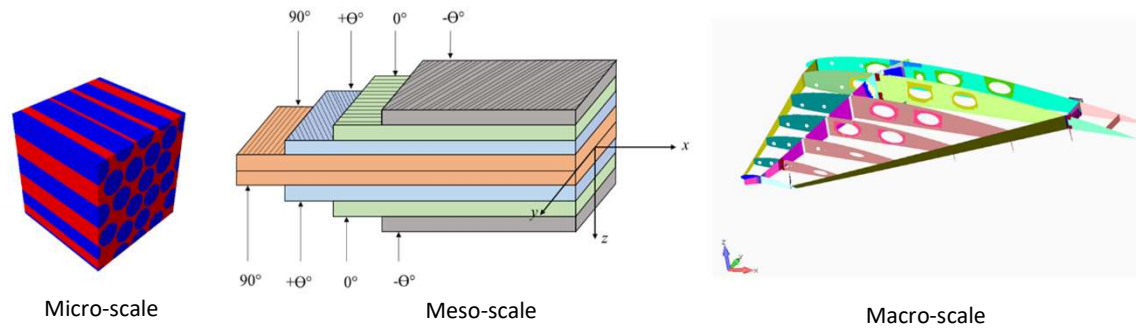


## Future Research Topics / Questions

- Thermoplastic Composites:  
Competition between Mechanisms: Discrete Damage and Plasticity
- OHT Modeling:  
Is it possible to predict thick laminates behavior without hi-fidelity modeling?  
ANN?  
Physics-Informed Reduced-Order Modeling?  
Double-Double (DD) Laminates with Homogenization
- Adaptive DNS-FE<sup>2</sup> modeling for Progressive Failure?
- Development of an Explicit Multi-Fidelity Model for Large Structures?

## Conclusion

### Multi-Fidelity Multi-Scale Modeling of Progressive Damage



Concurrent Direct FE<sup>2</sup>

Discrete Crack Methods

Smeared Crack Methods

Adaptivity

Coupled Models

Coupled Models

Towards Fundamental Properties

Towards Larger Structures

# Thank You



Contact: [mpetayte@nus.edu.sg](mailto:mpetayte@nus.edu.sg)



Department of Mechanical Engineering, National University of Singapore