

Liberté Égalité Fraternité



Monitoring of thermoset composites curing by SHM sensors

Presentation number 1451

ONERA – The French aerospace lab

M. Touraille, F. Saffar, P. Beauchêne, J-M. Roche

florence.saffar@onera.fr



Context: Control of the health state

Different methods for the control of structures:

→ NDT methods to control parts after the manufacturing step Ex: thermography pulse, ultrasonic methods



lightning strikes in UD composites detected by pulse thermography



Out-of-planes in composites detected by ultrasonic controls

 \rightarrow SHM methods:

mechanical impacts Detected by lock-in thermography

Embedment of sensors into the structure to monitor the health state and to trigger targeted maintenance operations



PZT sensors network on a stiffened panel



Lamb waves propagation in composites



Control of structure thermomechanical stresses by FBG

ightarrow Opportunity of the composites: Embed sensors into the core of the composite



Embedment of sensors into the core of the composites



Break of their integration:

Resistance of sensors to manufacturing conditions (temperature, pressure...)
Intrusiveness of sensors on composite properties



Embedment of a FBG into the core of a thermoset composite

Mechanical influence of the FBG embedment into the core of a composite

Goals/advantages:

- Getting knowledge of material's core behaviour
- Monitoring health state of material from manufacturing step
- > Freeing from bonding issues (bonding quality and repeatability, bonding aging...)
- Protection of sensors from environment conditions





Influence of the sensors debonding on their response

Definition of the system



Microscopic observation of the prepreg

Fibre Bragg Grating (FBG):

Optical sensor sensitive to temperatures and strains

$$\frac{\Delta \lambda_{\rm B}}{\lambda_{\rm B}} = K_{\rm T} \cdot \Delta T + K_{\rm \epsilon} \cdot \Delta \epsilon + K_{\rm P} \cdot \Delta P$$

 \succ Gauge length = 5mm

ONERA

THE FRENCH AEROSPACE LAB

RÉPUBLIQUE

FRANCAIŠE

Diameter = 250µm with acrylate coating, 125µm without coating





Characterisation of the prepreg curing kinetic (1/2)



RÉPUBLIQUE

THE FRENCH AEROSPACE LAB

FRANCAIŠE Égalité

Exaternit



Characterisation of the prepreg curing kinetic (2/2)

Identification of a curing kinetic law

Selection of a Kamal-Sourour model

$$\frac{d\alpha}{dt} = (k_1 + k_2 \alpha^m)(1 - \alpha)^n \quad k_i = A_i \exp(\frac{-E_{ai}}{RT})$$





Instrumentation of the manufacturing process



Instrumentation:

> 1 FBG in the core of the laminate at half-thickness, oriented perpendicular to the carbon fibres

> 2 monitoring thermocouples:

1 in the core of the laminate at half-thickness 1 on the top of the laminate





Preliminary results

THE FRENCH AEROSPACE LAB

Egalité Frateraité



Influence of process parameters on sensors responses

Evolution of curing parameters:

- > Influence of 3 parameters: Heating rate, temperature of the plateau, plateau duration
- Good tests repeatability:
 - slight differences during the heating phase BUT similar behaviour at the plateau and during the cooling phase
- Strain variations between 6000 and 8000 µdef (in function of the curing conditions)



Study of curing phenomena

ONERA

THE FRENCH AEROSPACE LAB

RÉPUBLIQUE

FRANÇAISE Liberté Égalité Fratteraité



CCM 23

Different identified steps:

Before curing
Low adhesion of the FBG
2a: Start of curing

Compensation of thermal expansion and chemical shrinkage

2b: Important chemical shrinkage

2c: Stabilisation of the chemical shrinkage

3: Thermal shrinkage

Evolution of the chemical shrinkage (1/2)

Strains and curing cycle:

For different manufacturing conditions (at every moment of the cycle):

- Strains measured by the FBG
- Curing degree estimated by the Kamal-Sourour model

Nonlinearity between the cure degree and the chemical shrinkage

Threshold value of the cure degree around 0.5





Evolution of the chemical shrinkage (2/2)

Study of the final chemical shrinkage:

For different manufacturing conditions (at the end of the plateau):

Maximum chemical shrinkage considered

THE FRENCH AEROSPACE LAB

RÉPUBLIQUE

FRANÇAIŠE Liberté Égalité Frateraité

Maximum curing degree estimated by the Kamal-Sourour model

CCM 23

Singular points: data dispersion and high chemical shrinkages

Hyp: Influence of TP nodules melting (T_{plateau} > T_{m nodules})







Conclusions and Perspectives

Conclusions:

- Reliable and repeatable monitoring of a thermoset composites curing by FBG
- Identification of the different curing phases
- Possibility to estimate the end of the curing from strains monitoring
- > Establishment of the link between the chemical shrinkage and the curing degree (non-linear)



Influence of chemical shrinkage on the composite coefficient of thermal expansion



Perspectives:

Improve the estimation of the curing degree from the strains measurement
Study the influence of thermoplastic nodules melting on the strains measurement
Validation of this method on another thermoset prepreg
Impact of curing degree on composite properties

Thank you for your attention!

Any question?

